

# UC Berkeley

## SEMM Reports Series

### Title

A General Framework for Continuum Damage Models, Part I: Infinitesimal Plastic Damage Models in Stress Space

### Permalink

<https://escholarship.org/uc/item/1h68w1ww>

### Authors

Armero, Francisco

Oller, Sergio

### Publication Date

1999-05-01

**REPORT NO.  
UCB/SEMM-1999/07**

**STRUCTURAL ENGINEERING,  
MECHANICS AND MATERIALS**

**A GENERAL FRAMEWORK FOR CONTINUUM  
DAMAGE MODELS.**

**PART I: INFINITESIMAL PLASTIC DAMAGE  
MODELS IN STRESS SPACE**

**BY**

**F. ARMERO**

**AND**

**S. OLLER**

**MAY 1999**

**DEPARTMENT OF CIVIL AND ENVIRONMENTAL ENGINEERING  
UNIVERSITY OF CALIFORNIA  
BERKELEY, CALIFORNIA**

# A General Framework for Continuum Damage Models. Part I: Infinitesimal Plastic Damage Models in Stress Space

by

FRANCISCO ARMERO\*

Structural Engineering, Mechanics and Materials  
Department of Civil and Environmental Engineering  
University of California, Berkeley CA 94720, USA

and

SERGIO OLLER<sup>‡</sup>

E.T.S. Ingenieros de Caminos, Canales y Puertos  
Universitat Politècnica de Catalunya  
Campus Nord, Modulo C1, 08034 Barcelona, Spain

## Abstract

We identify in this paper a general framework for the development of continuum damage models in their fully coupled plastic damage form. The focus of this first part is directed to the general formulation of infinitesimal models defined by yield and damage surfaces in stress space. The main feature of the proposed formulation is the direct and independent consideration of the damage mechanisms (isotropic damage, cracking,...) degrading the stiffness of the material, thus allowing for a complete physical characterization of these effects. This modular structure is accomplished by a kinematic decomposition of the strains in an elastic, plastic and multiple damage parts, as belonging to each activated damage mechanism. An additive decomposition in the infinitesimal range of interest is considered. Based on this decomposition, the constitutive characterization alluded to above for each damage mechanism is carried out in a complete thermodynamically consistent framework. One of the virtues of the considered framework is the fact that it includes many of the diverse damage models existing in the literature as particular cases. In this way, the developments presented herein furnish a unified framework for the formulation of continuum damage models, including isotropic damage, compliance based formulations, effective stress anisotropic models, smeared crack models and the related formulations of cracking and damage based on strong discontinuities. Besides the clear physical significance added to these existing formulations, the proposed framework defines also a very convenient context for the efficient numerical integration of the resulting models. This aspect is explored in Part II of this work, as it is the application of the framework proposed herein to the numerical simulation of porous metals.

**KEYWORDS:** continuum damage models, plasticity and damage.

---

\* Corresponding author.

<sup>‡</sup> On leave (Fall Semester 1998) in the Dept. of Civil and Envir. Engr., UC Berkeley.

## 1. Introduction

Damage in solids usually refers to the degradation of their elastic response upon unloading. In this way, the elastic moduli of the material decrease as further damage loading occurs. Added to these effects we can find the presence of plasticity characterized by the permanent strains in the material after full unloading. This observation has made very popular (almost universal) the formulation of plasticity models based on the kinematic decomposition of the strains in an elastic and a plastic or permanent parts. In contrast, the formulation of damage models has been based on much more diverse, and usually unrelated, frameworks.

For example, the concepts of effective stress and/or strain can be found in the literature in the development of damage models. These ideas can be traced back to the pioneering work of KACHANOV [1986] in the context of isotropic damage models. The resulting models consider a scalar variable measuring the ratio between damage and intact surfaces on which the stresses act, thus defining naturally the concept of effective stress as the equivalent stress acting on the intact material. More recent references, considering also plasticity effects thus leading to a coupled damage-plastic model, include SIMO & JU [1987], JU [1989], LUBLINER ET AL [1989] and LUCCIONI ET AL [1996], among many others. Extensions of these ideas to the anisotropic case leads to the consideration of rank four tensors, defined in terms of generically called “damage tensors”, relating the total and effective stresses; see e.g. CORDEBOIS & SIDOROFF [1982] and MURAKAMI [1983], involving rank two damage tensors. The characterization of the evolution of these damage tensors is usually very complex mathematically and difficult (if not impossible) to motivate physically.

In contrast, we can find more physically motivated approaches in the modeling of damage in concrete. Following the characterization of damage as the degradation of the elastic moduli, we can find in the literature a number of models considering the fourth order secant compliance of the material as fundamental internal variable in the formulation. The evolution of this compliance is then usually obtained in a thermodynamically consistent framework. Examples of this class of damage models can be found in ORTIZ [1985], SIMO & JU [1987], JU [1989], HANSEN & SCHREYER [1994] and GOVINDJEE, KAY & SIMO [1995], among others. Yet to incorporate more directly the physical mechanisms causing the damage in the material, namely the cracking of the concrete, we find models considering directly reduced stress-strain relations across the cracks. A classical example is furnished by the so-called smeared cracks models, as developed in the works of RASHID [1968], BAZANT & CEDOLIN [1979], BAZANT & OH [1983] and ROTS et al [1985], among other early references. Related to these ideas, we can also quote the so-called microplane approach of BAZANT & OH [1985], where the damage is introduced not through a particular crack direction but through a collection of predefined planes. Smeared crack models, however, have not been linked in the literature to the above developments in continuum damage theories. We have recently presented a formulation of these models in a completely

thermodynamic framework in the context of strong discontinuities; see ARMERO [1997a,b], and ARMERO & LI [1998] for preliminary results of its extension to the finite deformation range. This context provides, in addition, the proper multi-scale treatment for the regularization of the localized failure of the material in the large-scale models of interest; see ARMERO [1999].

We identify in the current paper a general framework for the formulation of continuum damage models that, in particular, unifies these different treatments found in the literature, including the smeared crack models. Our attention is directed to the formulation of infinitesimal plastic damage models defined by yield and damage surfaces in stress space. The fundamental observation in the proposed framework is to base all the developments on the kinematic decomposition of the strains in an elastic, plastic and damage parts, following ideas common to elastoplastic models. With the damage strain at hand, the dissipative mechanisms causing the damage in the material can readily be modeled, independent of the elastic and other responses in the material. This allows a complete physical interpretation of the resulting equations. For example, we consider an energy potential modeling the damage stress/strain relations associated to a damage mechanism (e.g. crack stress/displacement relations), including inelastic effects like hardening/softening cohesive laws. The presence of this damage potential identifies the recoverable nature of the damage strains. Special care is given to the identification of the different components of the stress and strain causing the damage in the material: volumetric, deviatoric, stress/strains associated to a given direction (crack), etc.

We show the extent of the proposed framework with the consideration of the particular simple case of a quadratic damage potential, involving the evolution of a set of reduced compliances associated only to the damage mechanism. We recover then as particular case the formulation presented in the seminal work of ORTIZ [1985] for the purely elastic-damage case. As shown in detail herein, the framework of smeared crack models is also recovered when the identified general framework is considered in its reduced form. It is interesting to observe that even though the consideration of damage strains (or in the form of “crack strains” as it is more common) can be found in these early references, their use in the formulation of continuum damage models has essentially disappeared in more recent literature. Instead, emphasis is given to the so-called “degrading strains”, while maintaining the total strains decomposed only in elastic and plastic parts; representative references of these considerations are the works DOUGILL [1976], HUECKEL & MAIER [1977], ORTIZ [1987] and YAZADANI & SCHREYER [1988], among others. The “degrading strains” are associated to the compliance degradation of the material (i.e.  $\dot{\mathbf{D}}\sigma$  for the rate of the compliance tensor  $\mathbf{D}$  and the stress tensor  $\sigma$ ). Therefore, they do not correspond to an actual “strain”, in particular a recoverable strain like the damage strains, since they are only defined through a rate when damage is active. As shown in the more recent work of CAROL et al [1994], the consideration of these “degrading strains” leads to a partial unification of several of the aforementioned approaches in damage modeling, leading to a

formal structure similar to elastoplastic models. We refer to this attempt as partial because models like the smeared crack approaches were not considered. In contrast, we show in this paper that the fundamental unifying assumption (in the theoretical developments and more especially at the level of the numerical integration of the resulting models, since the standard structure of return mapping algorithms common in elastoplastic models is recovered) is to be traced back to the elastic-plastic-damage decomposition of the total strains. Furthermore, it is the direct consideration of the damage strains and not the “degrading strains” that allows the physical characterization of the damage mechanisms.

We focus the developments in this first part on the consideration of plastic damage models in the infinitesimal case, with the threshold of damage characterized by surfaces in stress space. The general framework is developed in detail as it is the identification of existing models, giving the unifying character described above. We present in Part II the application of these ideas to the formulation of a simple plastic damage model of ductile failure in porous metals, including the effects of closing/opening of voids, an effect usually referred to as unilateral damage (see e.g. CHABOCHE [1995]). Numerical algorithms for the integration of the plastic damage framework investigated herein can also be found in Part II.

An outline of the rest of this paper is as follows. Section 2 develops the general framework proposed in this work. Emphasis is given to the incorporation of multiple damage mechanisms in the inelastic response of the material, hence illustrating the ability of the proposed formulation to handle the physical characterization of each of them independently. A completely consistent thermodynamic framework is considered for a particular damage mechanism. Section 3 particularizes the preceding developments to a quadratic damage potential. We show in the different subsections of Section 3.1 that this case includes as particular cases existing damage models, from isotropic damage to smeared crack models. Finally, Section 4 concludes with some final remarks.

## 2. General Formulation

We develop in this section the general framework for the formulation of damage theories proposed in this work. The fundamental kinematic assumptions underlying the proposed framework are presented in Section 2.1, with the resulting evolution equations of the considered internal variables described in Section 2.3 after deriving the dissipation expression in Section 2.2. The tangent rate equations are derived in Section 2.4.

### 2.1. Fundamental kinematic assumptions

With the assumption of infinitesimal deformations considered herein, the kinematics of the deformation at a given point  $\boldsymbol{x}$  of a solid  $\Omega$  (identified with its reference placement with

the above assumption) is characterized, under the classical assumption of local response, by the infinitesimal strain tensor

$$\boldsymbol{\varepsilon} := \text{sym}[\nabla \mathbf{u}] \in \mathcal{S}, \quad (2.1)$$

for the displacement field  $\mathbf{u} : \Omega \rightarrow \mathbb{R}^{n_{\text{dim}}}$  ( $n_{\text{dim}} = 1, 2$  or  $3$ ) and spatial gradient  $\nabla(\cdot)$ . We denote by  $\mathcal{S}$  the linear space of symmetric  $n_{\text{dim}} \times n_{\text{dim}}$  tensors. In this context, we introduce the fundamental decomposition of the total strain tensor  $\boldsymbol{\varepsilon}$  in

$$\boxed{\boldsymbol{\varepsilon} = \boldsymbol{\varepsilon}^e + \boldsymbol{\varepsilon}^p + \boldsymbol{\varepsilon}^d}, \quad (2.2)$$

that is, in elastic  $\boldsymbol{\varepsilon}^e$ , plastic  $\boldsymbol{\varepsilon}^p$  and damage  $\boldsymbol{\varepsilon}^d$  parts. The latter is assumed decomposed additively as

$$\boldsymbol{\varepsilon}^d = \sum_{d_I=1}^{n_{\text{dam}}} \boldsymbol{\varepsilon}^{d_I}, \quad (2.3)$$

that is, as the sum of  $n_{\text{dam}}$  similar terms.

Additive kinematic decompositions of the form (2.2) and (2.3) are commonly used in the context of elastoplastic theories, but are rarely considered in the context of damage models. It is the main goal of this paper to explore this decomposition in this latter case, and to show the convenience of their use from the point of view of the modeling of the damage response of the material and its subsequent numerical integration. In particular, the assumed decomposition (2.3) characterizes a part of the strain given by  $n_{\text{dam}}$  “damage mechanisms”. More specifically, it corresponds to a decomposition of the deformation at a given point of the solid in elastic and damage mechanisms *in series*, referring to the usual convention in a generalized standard solid. In contrast with the plastic part  $\boldsymbol{\varepsilon}^p$ , the damage strain  $\boldsymbol{\varepsilon}^d$  is thought as a recoverable strain as it will be apparent by its role in the stored free energy of the material.

As illustrated in Section 3 below, the damage mechanisms characteristic of the response of many materials lead to particular constrained forms of the associated damage strains  $\boldsymbol{\varepsilon}^d$ ; typical examples involve volumetric damage strains, rank-one strains associated to cracks in brittle materials, etc. In general, and for each damage mechanism ( $d_I = 1, n_{\text{dam}}$ ), we have  $\boldsymbol{\varepsilon}^{d_I} \in \mathcal{V}^{d_I}$  for the linear subspace

$$\mathcal{V}^{d_I} = \left\{ \boldsymbol{\varepsilon}^{d_I} = \sum_{a=1}^{n_{d_I}} \mathbb{P}_a^{d_I} e_a^{d_I} \quad \text{for } e_a^{d_I} \in \mathbb{R} \quad \text{and} \quad \mathbb{P}_a^{d_I} \in \mathcal{S} \right\} \subset \mathcal{S}, \quad (2.4)$$

where the set of symmetric tensors  $\{\mathbb{P}_a^{d_I}\}_{a=1}^{n_{d_I}}$  is assumed to define a basis of  $\mathcal{V}^{d_I}$ , with dimension  $n_{d_I} \leq (n_{\text{dim}} + 1)n_{\text{dim}}/2$ . Without loss of generality, we consider an orthonormal basis in the usual inner product of symmetric tensors  $\mathcal{S}$  (that is, double contraction  $\mathbf{A} : \mathbf{A} := A_{ij}A_{ij}$ , summation implied). In this way, we have the orthogonality relations

$$\mathbb{P}_a^{d_I} : \mathbb{P}_b^{d_I} = \delta_{ab} \quad (\text{Kronecker delta}), \quad (2.5)$$

and the component relations

$$e_a^{d_I} = \mathbb{P}_a^{d_I} : \boldsymbol{\varepsilon}^{d_I} \quad \text{for } a = 1, n_{d_I}, \quad (2.6)$$

for each damage mechanism  $d_I = 1, n_{dam}$  independently. To avoid the use of double indices when referring to the damage mechanism and damage strain component, we write the component relation in (2.4) in the compact form

$$\boldsymbol{\varepsilon}^{d_I} = \mathbb{P}^{d_I T} \mathbf{e}^{d_I} \quad \text{for} \quad \mathbf{e}^{d_I} := [e_a^{d_I}] = \begin{bmatrix} e_1^{d_I} \\ \vdots \\ e_{n_{d_I}}^{d_I} \end{bmatrix}, \quad (2.7)$$

and  $\mathbb{P}^{d_I} = [\mathbb{P}_{a(ij)}^{d_I}]$  for  $a = 1, n_{d_I}$  and  $i, j = 1, n_{\text{dim}}$ , with  $\mathbb{P}^{d_I T} := [\mathbb{P}_{(ij)a}^{d_I}]$ . We write symbolically

$$\mathbb{P}^{d_I} = \begin{bmatrix} \mathbb{P}_1^{d_I} \\ \vdots \\ \mathbb{P}_{n_{d_I}}^{d_I} \end{bmatrix}, \quad (2.8)$$

in the developments that follow.

### Remarks 2.1.

1. In the numerical implementation, we write the strains  $\boldsymbol{\varepsilon}$  and stresses  $\boldsymbol{\sigma}$  in the standard vector notation

$$\boldsymbol{\varepsilon} \rightarrow \begin{bmatrix} \varepsilon_{11} \\ \varepsilon_{22} \\ \varepsilon_{33} \\ 2\varepsilon_{12} \\ 2\varepsilon_{13} \\ 2\varepsilon_{32} \end{bmatrix} \quad \text{and} \quad \boldsymbol{\sigma} \rightarrow \begin{bmatrix} \sigma_{11} \\ \sigma_{22} \\ \sigma_{33} \\ \sigma_{12} \\ \sigma_{13} \\ \sigma_{32} \end{bmatrix}, \quad (2.9)$$

in general three-dimensional problems. In this notation,  $\mathbb{P}^{n_{d_I}}$  corresponds to a 6  $n_{d_I}$  vector (4  $n_{d_I}$  in plane problems).

2. We have assumed, for simplicity, that the projection matrices  $\{\mathbb{P}_a^{d_I}\}_{a=1}^{n_{d_I}}$  are constant. A general definition of  $\mathcal{V}^d$  as a nonlinear manifold in  $\mathcal{S}$  can be considered, with the space of variations given by the linear basis  $\{\mathbb{P}_a^{d_I}\}_{a=1}^{n_{d_I}}$  at a given point  $\mathbf{e}^d$ . To avoid the added complexity introduced by this general treatment, we consider only the simple case furnished by a linear space of damage strains. The case of a nonlinear definition of the damage space is treated separately by imposing explicitly the constraints. This is done in Part II of this work (ARMERO & OLLER [1999]) for the imposition of unilateral damage associated to the closing/opening of voids in a particular coupled plastic damage model of porous metals.  $\square$

## 2.2. Elastic energy and plastic-damage dissipation

The considerations described in the previous section identified the kinematics of the problem of interest. In particular, we considered the existence of  $n_{dam}$  mechanisms leading to the damage of the solid. In accordance, we introduce the decomposition of the stored energy function

$$W = W^e(\boldsymbol{\varepsilon}^e) + \mathcal{H}^p(\mathcal{I}^p) + \sum_{d_I=1}^{n_{dam}} W^{d_I}(\boldsymbol{\varepsilon}^{d_I}; \mathcal{I}^{d_I}), \quad (2.10)$$

for a stored energy function  $W^e(\cdot)$  characterizing the elastic response of the solid in terms of the elastic strains  $\boldsymbol{\varepsilon}^e$ , a potential  $\mathcal{H}^p(\cdot)$  modeling the hardening plastic response in terms of a general set of plastic internal variables  $\mathcal{I}^p$  (not including  $\boldsymbol{\varepsilon}^p$ ), and a set of stored energy functions  $W^{d_I}(\cdot)$  characterizing the internal energy stored in each of the  $n_{dam}$  damage mechanisms. The latter are given in terms of the corresponding damage strain  $\boldsymbol{\varepsilon}^{d_I}$  and a general set of internal variables  $\mathcal{I}^{d_I}$  (not including  $\boldsymbol{\varepsilon}^{d_I}$ ) characterizing the inelastic response associated to the damage mechanisms. For simplicity, we assume a full decoupling of the different damage mechanisms, leading to an independent set of internal variables for each of them. We note from the start that the presence of the damage strains  $\boldsymbol{\varepsilon}^d$  explicitly in the expression of the stored energy function, in contrast with the plastic strains  $\boldsymbol{\varepsilon}^p$ , leads to their recoverable character, as discussed below. Uncoupled thermomechanical conditions have been assumed, with the identification of the stored energy function  $W$  with the Helmholtz free energy of the material in the case of isothermal deformations as it is customary.

Given the expression (2.10) of the stored energy of the material, the local dissipation rate (see e.g. SIMO & HUGHES [1998]) can be written, after using the kinematic decomposition (2.2), as

$$\begin{aligned} \mathcal{D} &= \boldsymbol{\sigma} : \dot{\boldsymbol{\varepsilon}} - \dot{W} \\ &= \left( \boldsymbol{\sigma} - \frac{\partial W}{\partial \boldsymbol{\varepsilon}^e} \right) : \dot{\boldsymbol{\varepsilon}}^e + \underbrace{\left( \boldsymbol{\sigma} : \dot{\boldsymbol{\varepsilon}}^p - \dot{\mathcal{H}}^p \right)}_{\mathcal{D}^p} + \sum_{d_I=1}^{n_{dam}} \underbrace{\left( \boldsymbol{\sigma} : \dot{\boldsymbol{\varepsilon}}^{d_I} - \dot{W}^{d_I} \right)}_{\mathcal{D}^{d_I}} \geq 0, \end{aligned} \quad (2.11)$$

for the stress tensor  $\boldsymbol{\sigma}$ . After imposing the physically motivated constraints

$$\mathcal{D}^p \geq 0 \quad \text{and} \quad \mathcal{D}^{d_I} \geq 0 \quad d_I = 1, n_{dam}, \quad (2.12)$$

for the plastic and damage dissipations, the imposition of the physical inequality  $\mathcal{D} \geq 0$  for all elastic strain rates  $\dot{\boldsymbol{\varepsilon}}^e$  leads to the constitutive equation

$$\boldsymbol{\sigma} = \frac{\partial W^e}{\partial \boldsymbol{\varepsilon}^e}. \quad (2.13)$$

The procedure outlined here follows the classical arguments known as Coleman's method (see e.g. TRUESDELL & NOLL [1965] and LUBLINER [1990]).

The plastic dissipation in (2.11) can be written as

$$\boxed{\mathcal{D}^p = \boldsymbol{\sigma} : \dot{\boldsymbol{\varepsilon}}^p + Q^p \odot \dot{\mathcal{I}}^p \geq 0 \quad \text{for} \quad Q^p := -\frac{\partial \mathcal{H}^p}{\partial \mathcal{I}^p},} \quad (2.14)$$

the conjugate plastic hardening variable. The symbol  $\odot$  in this last expression denotes a general tensor contraction depending on the character of the internal variables  $\mathcal{I}^p$ . The specific case of a scalar variable  $\mathcal{I}^p \equiv \alpha^p$ , with scalar conjugate variable  $Q^p \equiv q^p$  and standard scalar multiplication between them, is considered in the particular model example of Section 3.

The damage dissipation  $\mathcal{D}^{d_I}$  in (2.11) can be expressed in the same way by

$$\mathcal{D}^{d_I} = \left( \boldsymbol{\sigma} - \frac{\partial W^{d_I}}{\partial \boldsymbol{\varepsilon}^{d_I}} \right) : \dot{\boldsymbol{\varepsilon}}^{d_I} - \frac{\partial W^{d_I}}{\partial \mathcal{I}^{d_I}} \odot \dot{\mathcal{I}}^{d_I} \geq 0, \quad (2.15)$$

where  $\odot$  denotes again a general contraction depending on the character of the internal variables  $\mathcal{I}^{d_I}$ . Following the same arguments as before, that is, imposing the non-negative character of the dissipation (2.15) for independent variations of the damage strains  $\boldsymbol{\varepsilon}^{d_I}$  and internal variables  $\mathcal{I}^{d_I}$ , leads after noting the constrained character of (2.15) (since  $\boldsymbol{\varepsilon}^{d_I} \in \mathcal{V}^{d_I}$ ) to

$$\boxed{\mathcal{D}^{d_I} = Q^{d_I} \odot \dot{\mathcal{I}}^{d_I} \geq 0 \quad \text{for} \quad Q^{d_I} := -\frac{\partial W^{d_I}}{\partial \mathcal{I}^{d_I}},} \quad (2.16)$$

and the general relation

$$\boldsymbol{\sigma} = \frac{\partial W^{d_I}}{\partial \boldsymbol{\varepsilon}^{d_I}} + \boldsymbol{\sigma}^\perp, \quad (2.17)$$

for a component  $\boldsymbol{\sigma}^\perp \in \mathcal{V}^{d_I \perp}$  defined by

$$\mathcal{V}^{d_I \perp} := \left\{ \boldsymbol{\sigma} \in \mathcal{S} : \boldsymbol{\sigma} : \boldsymbol{\varepsilon}^{d_I} = 0 \quad \forall \boldsymbol{\varepsilon}^{d_I} \in \mathcal{V}^{d_I} \right\}, \quad (2.18)$$

that is, the orthogonal complement of the original space of damage strains  $\mathcal{V}^{d_I}$ . The explicit absence of the damage strains  $\boldsymbol{\varepsilon}^{d_I}$  in the expression of the damage dissipation (2.16)<sub>1</sub> emphasizes once more the recoverable character of these strains, in contrast with  $\boldsymbol{\varepsilon}^p$ .

If  $\boldsymbol{\sigma}^{d_I}$  denotes the projection of the a general stress tensor  $\boldsymbol{\sigma} \in \mathcal{S}$  onto the space  $\mathcal{V}^{d_I}$ , we can write the general relation

$$\boldsymbol{\sigma}^{d_I} = \sum_{a=1}^{n_{d_I}} \mathbb{P}_a^{d_I} s_a^{d_I} \quad \text{for} \quad s_a^{d_I} := \mathbb{P}_a^{d_I} : \boldsymbol{\sigma}, \quad (2.19)$$

or, simply,

$$\boldsymbol{\sigma}^{d_I} = \mathbb{P}^{d_I T} \mathbf{s}^{d_I} \quad \text{for} \quad \mathbf{s}^{d_I} = \begin{bmatrix} s_1^{d_I} \\ \vdots \\ s_{n_{d_I}}^{d_I} \end{bmatrix} = \mathbb{P}^{d_I} : \boldsymbol{\sigma}, \quad (2.20)$$

in the compact notation introduced in (2.7). We note that with this notation at hand the following equality holds

$$\boldsymbol{\sigma}^{d_I} : \boldsymbol{\varepsilon}^{d_I} = \mathbf{s}^{d_I} \cdot \mathbf{e}^{d_I} \quad \forall \boldsymbol{\sigma}^{d_I} = \mathbb{P}^{d_I T} \mathbf{s}^{d_I}, \quad \boldsymbol{\varepsilon}^{d_I} = \mathbb{P}^{d_I T} \mathbf{e}^{d_I} \in \mathcal{V}^{d_I}, \quad (2.21)$$

given the orthogonality relation (2.5). Projecting the relation (2.17) on  $\mathcal{V}^{d_I}$ , we have

$$\boldsymbol{\sigma}^{d_I} = \frac{\partial W^{d_I}}{\partial \boldsymbol{\varepsilon}^{d_I}}, \quad (2.22)$$

or, simply,

$$\boxed{\mathbf{s}^{d_I} = \frac{\partial \widehat{W}^{d_I}}{\partial \mathbf{e}^{d_I}}} \quad \text{or, in components,} \quad s_a^{d_I} = \frac{\partial \widehat{W}^{d_I}}{\partial e_a^{d_I}} \quad (a = 1, n_{d_I}), \quad (2.23)$$

where

$$\widehat{W}^{d_I}(\mathbf{e}^{d_I}; \mathcal{I}^{d_I}) = W^{d_I}(\boldsymbol{\varepsilon}^{d_I}; \mathcal{I}^{d_I}) \quad \text{for} \quad \boldsymbol{\varepsilon}^{d_I} = \mathbb{P}^{d_I T} \mathbf{e}^{d_I} \in \mathcal{V}^{d_I}, \quad (2.24)$$

after using the compact notation introduced in (2.7). We also note the trivial relation

$$\mathcal{Q}^{d_I} = -\frac{\partial W^d}{\partial \mathcal{I}^{d_I}} = -\frac{\partial \widehat{W}^d}{\partial \mathcal{I}^{d_I}}, \quad (2.25)$$

for future use.

### 2.3. The plastic damage evolution equations

The above thermodynamic framework leads directly to the evolution equations of the damage internal variables  $\mathcal{I}^{d_I}$  and plastic variables  $\{\boldsymbol{\varepsilon}^p, \mathcal{I}^p\}$ . In particular, we note from the expression (2.11) of the dissipation the conjugate character of the internal variables  $\mathcal{Q}^{d_I}$  and  $\mathcal{I}^{d_I}$  for the damage part and  $\{\boldsymbol{\varepsilon}^p, \mathcal{I}^p\}$  with  $\{\boldsymbol{\sigma}, \mathcal{Q}^p\}$  for the plastic part. In this context, we introduce a set of damage surfaces  $\phi^{d_I}(\mathcal{Q}^{d_I})$  for each damage mechanism  $d_I = 1, n_{dam}$  and a plastic yield surface  $\phi^p(\boldsymbol{\sigma}, q)$  by the relations

$$\phi^{d_I}(\mathcal{Q}^{d_I}) \leq 0 \quad \text{and} \quad \phi^p(\boldsymbol{\sigma}, q) \leq 0, \quad (2.26)$$

defining the elastic domain of the materials response.

A classical argument based on the stationarity of the dissipation functionals (2.14) and (2.16)<sub>1</sub> leads to the associated damage evolution equations for the damage and plastic internal variables; and convexity requirements for the damage and yield criteria; see e.g. LUBLINER [1990] and SIMO & HUGHES [1998] for similar arguments in the context of elastoplasticity. In particular, the damage evolution equations read

$$\boxed{\dot{\mathcal{I}}^{d_I} = \gamma^{d_I} \frac{\partial \phi^{d_I}}{\partial \mathcal{Q}^{d_I}}}, \quad (2.27)$$

for the damage multiplier  $\gamma^{d_I}$  satisfying the complementary Kuhn-Tucker loading/unloading conditions

$$\gamma^{d_I} \geq 0, \quad \phi^{d_I} \leq 0 \quad \text{and} \quad \gamma^{d_I} \phi^{d_I} = 0, \quad (2.28)$$

as well as the damage consistency condition

$$\gamma^{d_I} \dot{\phi}^{d_I} = 0, \quad (2.29)$$

during persistent damage. Similarly, the associated plastic evolution equations read

$$\boxed{\dot{\epsilon}^p = \gamma^p \frac{\partial \phi^p}{\partial \sigma}, \quad \dot{I}^p = \gamma^p \frac{\partial \phi^p}{\partial Q^p}}, \quad (2.30)$$

for the plastic multiplier  $\gamma^p$  satisfying

$$\gamma^p \geq 0, \quad \phi^p \leq 0 \quad \text{and} \quad \gamma^p \phi^p = 0, \quad (2.31)$$

and the plastic consistency condition

$$\gamma^p \dot{\phi}^p = 0. \quad (2.32)$$

The above relations define completely the damage model. A summary of these equations can be found in Box 2.1.

## Remarks 2.2.

1. Additional extensions can be easily accommodated in the above framework. For example, the case of multi-surface damage mechanisms can be easily considered through a damage evolution equation of the Koiter type (see e.g. SIMO & HUGHES [1998]) as

$$\dot{I}^{d_I} = \sum_{j=1}^{n_{surf}^{d_I}} \gamma_j^{d_I} \frac{\partial \phi_j^{d_I}}{\partial Q^{d_I}}, \quad (2.33)$$

$$\gamma_j^{d_I} \geq 0, \quad \phi_j^{d_I} \leq 0, \quad \sum_{j=1}^{n_{surf}^{d_I}} \gamma_j^{d_I} \phi_j^{d_I} = 0 \quad \text{and} \quad \sum_{j=1}^{n_{surf}^{d_I}} \gamma_j^{d_I} \dot{\phi}_j^{d_I} = 0, \quad (2.34)$$

for  $j = 1, n_{surf}^{d_I}$  damage surfaces associated to the damage mechanism  $d_I = 1, n_{dam}$ .

2. Similarly, the formulation of viscous damage models is easily accomplished by the viscous regularization of the Perzyna type, namely, replacing the Kuhn-Tucker loading and unloading conditions (2.28) and consistency condition (2.29) by the evolution equation

$$\gamma^{d_I} = \frac{\langle f^{d_I}(\phi^{d_I}) \rangle}{\eta^{d_I}}, \quad (2.35)$$

for a damage viscosity  $\eta^{d_I}$ , general scalar function  $f^{d_I}(\cdot)$ , and Macaulay brackets

$$\langle x \rangle := \begin{cases} 0 & \text{if } x \leq 0, \\ x & \text{if } x \geq 0, \end{cases} \quad (2.36)$$

while retaining the evolution equations (2.27). Similar considerations apply to the viscoplastic case.  $\square$

### 2.3.1. Damage models in stress space

The above arguments led to the consideration of damage surfaces defined in terms of the internal variables  $\mathcal{Q}^{d_I}$ . These variables are defined by (2.16)<sub>2</sub> and, thus, they can be written as

$$\mathcal{Q}^{d_I} := -\frac{\partial \widehat{W}^{d_I}}{\partial \mathcal{I}^{d_I}} = \widehat{\mathcal{Q}}^{d_I}(e^{d_I}; \mathcal{I}^{d_I}) \quad (2.37)$$

that is, as functions of the damage strains  $e^{d_I}$  (or, equivalently,  $\varepsilon^{d_I}$ ) and conjugate internal variables  $\mathcal{I}^{d_I}$ . However, and as noted in the introduction, our interest in this paper is the consideration of stress-based damage models, that is, with the damage surfaces  $\phi^{d_I}$  in terms of the stresses  $\sigma$ . To this purpose, the internal variables  $\mathcal{Q}^{d_I}$  can be alternatively defined as follows.

We introduce the *damage complementary energy function*, denoted for the mechanism  $d_I = 1, n_{dam}$  by

$$\chi^{d_I}(\sigma^{d_I}; \mathcal{I}^{d_I}) = \widehat{\chi}^{d_I}(s^{d_I}; \mathcal{I}^{d_I}) \quad \text{for } \sigma^{d_I} = \mathbb{P}^{d_I T} s^{d_I} \in \mathcal{V}^{d_I}, \quad (2.38)$$

and obtained by the Legendre transform of the stored energy function  $W^{d_I}$  in the damage strains. That is, for a given stress  $\sigma^{d_I} = \mathbb{P}^{d_I T} s^{d_I} \in \mathcal{V}^{d_I}$ , we define the function

$$\widehat{\chi}^d(s^{d_I}; \mathcal{I}^{d_I}) := \max_{e^{d_I}} \left\{ \underbrace{s^{d_I} \cdot e^{d_I}}_{\sigma^{d_I} : \varepsilon^{d_I}} - \widehat{W}^{d_I}(e^{d_I}; \mathcal{I}^{d_I}) \right\} \quad (2.39)$$

$$= s^{d_I} : \widehat{e}^{d_I}(s^{d_I}, \mathcal{I}^{d_I}) - \widehat{W}^{d_I}(\widehat{e}^{d_I}(s^{d_I}, \mathcal{I}^{d_I}); \mathcal{I}^{d_I}), \quad (2.40)$$

for the function  $\widehat{e}^{d_I}(s^{d_I}, \mathcal{I}^{d_I})$  obtained by inverting the constitutive relation

$$s^{d_I} = \frac{\partial \widehat{W}^{d_I}}{\partial e^{d_I}}(e^{d_I}; \mathcal{I}^{d_I}). \quad (2.41)$$

A simple argument based on the chain rule applied to (2.40) shows that

$$\left. \frac{\partial \widehat{\chi}^{d_I}}{\partial s^{d_I}} \right|_{\mathcal{I}^{d_I}} = e^{d_I} + s^{d_I} : \underbrace{\left. \frac{\partial \widehat{e}^{d_I}}{\partial s^{d_I}} \right|_{\mathcal{I}^{d_I}} - \frac{\partial \widehat{W}^{d_I}}{\partial e^{d_I}} \bigg|_{\mathcal{I}^{d_I}}}_{s^{d_I}} : \left. \frac{\partial \widehat{e}^{d_I}}{\partial s^{d_I}} \right|_{\mathcal{I}^{d_I}} \quad (2.42)$$

so

$$\boxed{e^{d_I} = \frac{\partial \widehat{\chi}^{d_I}}{\partial \mathbf{s}^{d_I}}} \quad \text{or, in components,} \quad e_a^{d_I} = \frac{\partial \widehat{\chi}^{d_I}}{\partial s_a^{d_I}} \quad (a = 1, n_{d_I}), \quad (2.43)$$

where the arguments of the function  $\widehat{\chi}^{d_I}(\mathbf{s}^{d_I}; \mathcal{I}^{d_I})$  are implied. Equivalently, we write

$$\boldsymbol{\varepsilon}^{d_I} = \frac{\partial \chi^{d_I}}{\partial \boldsymbol{\sigma}^{d_I}}, \quad (2.44)$$

for  $\chi^{d_I}(\boldsymbol{\sigma}^{d_I}; \mathcal{I}^{d_I})$  defined in (2.38).

A calculation similar to (2.42) leads easily to the relation

$$\left. \frac{\partial \widehat{\chi}^{d_I}}{\partial \mathcal{I}^{d_I}} \right|_{\mathbf{s}^{d_I}} = - \left. \frac{\partial \widehat{W}^{d_I}}{\partial \mathcal{I}^{d_I}} \right|_{\mathbf{e}^{d_I}}, \quad (2.45)$$

thus defining the internal variables

$$\mathcal{Q}^{d_I} = \widehat{\mathcal{Q}}^{d_I}(\mathbf{s}^{d_I}; \mathcal{I}^{d_I}) = \frac{\partial \widehat{\chi}^{d_I}}{\partial \mathcal{I}^{d_I}}(\mathbf{s}^{d_I}; \mathcal{I}^{d_I}), \quad (2.46)$$

in terms of the stresses  $\mathbf{s}^{d_I}$  (or, equivalently,  $\boldsymbol{\sigma}^{d_I} = \mathbb{P}^{d_I T} \mathbf{s}^{d_I}$ ) associated to the damage mechanism  $d_I$ . The damage surfaces (2.26) can then be expressed as

$$\widehat{\phi}(\mathbf{s}^{d_I}; \mathcal{I}^{d_I}) := \phi^{d_I}(\widehat{\mathcal{Q}}^{d_I}(\mathbf{s}^{d_I}; \mathcal{I}^{d_I})), \quad (2.47)$$

in terms of the damage stresses  $\mathbf{s}^{d_I}$  and the conjugate internal variables  $\mathcal{I}^{d_I}$ . Box 2.1 summarizes the equation of the final plastic damage model.

### Remarks 2.3.

1. A sufficient condition for the invertibility of the relation (2.41) is the convexity of the damage potentials  $\widehat{W}^{d_I}(\mathbf{e}^{d_I}; \mathcal{I}^{d_I})$  in the damage strain argument  $\mathbf{e}^{d_I}$  (that is, for fixed  $\mathcal{I}^{d_I}$ ), which also implies the convexity of the complementary functions  $\chi^{d_I}$  in the stresses  $\mathbf{s}^{d_I}$ . This convexity applies for the cases of interest described in Section 3 below, and it is assumed hereafter, *once the damage mechanism is activated*.
2. The construction of the damage linear subspace  $\mathcal{V}^{d_I}$  is, in fact, motivated by the final relation (2.47) defining the damage surfaces in terms of the specific damage stress components  $\mathbf{s}^{d_I}$ . □

**BOX 2.1.** Summary of the constitutive relations for a general damage model

1. Elastic stress-strain relation

$$\boldsymbol{\sigma} = \frac{\partial W^e}{\partial \boldsymbol{\varepsilon}^e},$$

for an elastic potential  $W^e(\boldsymbol{\varepsilon}^e)$ .

2. Damage stress-strain relations

$$\boldsymbol{\sigma}^{d_I} = \frac{\partial W^{d_I}}{\partial \boldsymbol{\varepsilon}^{d_I}} \quad \text{or} \quad \boldsymbol{\varepsilon}^{d_I} = \frac{\partial \chi^{d_I}}{\partial \boldsymbol{\sigma}^{d_I}} \quad \left( \Leftrightarrow \quad \mathbf{s}^{d_I} = \frac{\partial \widehat{W}^{d_I}}{\partial \mathbf{e}^{d_I}} \quad \text{or} \quad \mathbf{e}^{d_I} = \frac{\partial \widehat{\chi}^{d_I}}{\partial \mathbf{s}^{d_I}} \right),$$

for the conjugate damage potentials  $W^{d_I}(\boldsymbol{\varepsilon}^{d_I}; \mathcal{I}^{d_I})$  and  $\chi^{d_I}(\boldsymbol{\sigma}^{d_I}; \mathcal{I}^{d_I})$  (or, equivalently,  $\widehat{W}^{d_I}(\mathbf{e}^{d_I}; \mathcal{I}^{d_I})$  and  $\widehat{\chi}^{d_I}(\mathbf{s}^{d_I}; \mathcal{I}^{d_I})$ ), for each damage mechanism  $d_I = 1, n_{dam}$ .

The damage stresses  $\boldsymbol{\sigma}^{d_I}$  are given in terms of the (total) stresses  $\boldsymbol{\sigma}$  by the projection relation (equilibrium)

$$\boldsymbol{\sigma}^{d_I} = \sum_{a=1}^{n_{d_I}} \mathbb{P}_a^{d_I} \mathbf{s}_a^{d_I} \quad \text{with} \quad \mathbf{s}_a^{d_I} = \mathbb{P}_a^{d_I} : \boldsymbol{\sigma},$$

between the two sets of stresses.

3. The (associated) damage evolution equations

$$\left. \begin{aligned} \dot{\mathcal{I}}^{d_I} &= \gamma^{d_I} \frac{\partial \phi^{d_I}}{\partial \mathcal{Q}^{d_I}}, \\ \gamma^{d_I} &\geq 0, \quad \phi^{d_I} \leq 0, \quad \gamma^{d_I} \phi^{d_I} = 0, \\ \gamma^{d_I} \dot{\phi}^{d_I} &= 0, \end{aligned} \right\}$$

where  $\mathcal{Q}^{d_I} = -\frac{\partial \widehat{W}^{d_I}}{\partial \mathcal{I}^{d_I}} = \frac{\partial \widehat{\chi}^{d_I}}{\partial \mathcal{I}^{d_I}}$ , for each damage mechanism  $d_I = 1, n_{dam}$ .

4. The (associated) damage evolution equations

$$\left. \begin{aligned} \dot{\boldsymbol{\varepsilon}}^p &= \gamma^p \frac{\partial \phi^p}{\partial \boldsymbol{\sigma}}, \quad \dot{\mathcal{I}}^p = \gamma^p \frac{\partial \phi^p}{\partial \mathcal{Q}^p}, \\ \gamma^p &\geq 0, \quad \phi^p \leq 0, \quad \gamma^p \phi^p = 0, \\ \gamma^p \dot{\phi}^p &= 0. \end{aligned} \right\}$$

where  $\mathcal{Q}^p = -\frac{\partial \mathcal{H}^p}{\partial \mathcal{I}^p}$ .

## 2.4. The rate equations

To identify better the damage response introduced by the considerations presented in the previous sections by the total stress-strain relations, we derive in this section the tangent rate equations associated to the general models developed above. As noted in the introduction, one of the main advantages of the proposed formulation is the modularity in the treatment of simultaneous damage mechanisms. We can exploit this modularity by first deriving the tangent rate form of the equations for each of the damage mechanisms to account for their combined action together with the elastoplastic response of the material afterwards. In this way, we consider a given damage mechanism  $d_I = 1, n_{dam}$  and derive the rate equation

$$\dot{\mathbf{s}}^{d_I} = \mathbf{c}_{lu}^{d_I} \dot{\mathbf{e}}^{d_I} \quad \text{or, in components,} \quad \dot{s}_a^{d_I} = \sum_{b=1}^{n_{d_I}} c_{lu_{ab}}^{d_I} \dot{e}_b^{d_I} \quad (a = 1, n_{d_I}), \quad (2.48)$$

for a tangent matrix  $\mathbf{c}_{lu}^{d_I} \in \mathbb{R}^{n_{d_I} \times n_{d_I}}$  to be found, accounting for the active damage loading or inactive damage unloading state of the damage mechanism  $d_I$ . We note the use in (2.48) of the reduced arrays in  $\mathcal{V}^{d_I}$ , thus avoiding in the following developments the singularities associated with the components of the strain and stress not involved in the damage of the material.

### 2.4.1. The tangent damage rate relations

The rate form of equation (2.43) reads

$$\begin{aligned} \dot{\mathbf{e}}^{d_I} &= \frac{\partial^2 \widehat{\chi}^{d_I}}{\partial \mathbf{s}^{d_I 2}} \dot{\mathbf{s}}^{d_I} + \frac{\partial^2 \widehat{\chi}^{d_I}}{\partial \mathbf{s}^{d_I} \partial \mathcal{I}^{d_I}} \odot \dot{\mathcal{I}}^{d_I} \\ &= \mathbf{d}^{d_I} \dot{\mathbf{s}}^{d_I} + \gamma^{d_I} \frac{\partial^2 \widehat{\chi}^{d_I}}{\partial \mathbf{s}^{d_I} \partial \mathcal{I}^{d_I}} \odot \frac{\partial \phi^{d_I}}{\partial Q^{d_I}}, \end{aligned} \quad (2.49)$$

where we have used the evolution equation (2.27) and introduced the damage compliance  $\mathbf{d}^{d_I}$  through the definition

$$\boxed{\mathbf{d}^{d_I} := \frac{\partial^2 \widehat{\chi}^{d_I}}{\partial \mathbf{s}^{d_I 2}}}. \quad (2.50)$$

The symmetry of the compliance  $\mathbf{d}^{d_I}$  follows. As noted in Remark 2.3.1, we assume the convexity of the damage potential  $\widehat{\chi}(\mathbf{s}^{d_I}; \mathcal{I}^{d_I})$  in the stress component, hence resulting in the positive definiteness of the compliance  $\mathbf{d}^{d_I}$  in (2.50) (and thus invertible), *once the damage mechanism  $d_I$  is activated*. We introduce the notation

$$\mathbf{c}^{d_I} := \mathbf{d}^{d_I^{-1}}, \quad (2.51)$$

for the damage tangent. Similarly, the rate form of the equation (2.46) leads to the relation

$$\begin{aligned}\dot{Q}^{d_I} &= \frac{\partial^2 \widehat{\chi}^{d_I}}{\partial \mathcal{I}^{d_I} \partial \mathbf{s}^{d_I}} \dot{\mathbf{s}}^{d_I} + \frac{\partial^2 \widehat{\chi}^{d_I}}{\partial \mathcal{I}^{d_I 2}} \odot \dot{\mathcal{I}}^{d_I} \\ &= \frac{\partial^2 \widehat{\chi}^{d_I}}{\partial \mathcal{I}^{d_I} \partial \mathbf{s}^{d_I}} \dot{\mathbf{s}}^{d_I} + \gamma^{d_I} \frac{\partial^2 \widehat{\chi}^{d_I}}{\partial \mathcal{I}^{d_I 2}} \odot \frac{\partial \phi^{d_I}}{\partial Q^{d_I}}\end{aligned}\quad (2.52)$$

$$\begin{aligned}&= \frac{\partial^2 \widehat{\chi}^{d_I}}{\partial \mathcal{I}^{d_I} \partial \mathbf{s}^{d_I}} \mathbf{c}^{d_I} \dot{\mathbf{e}}^{d_I} \\ &\quad - \gamma^{d_I} \left[ \frac{\partial^2 \widehat{\chi}^{d_I}}{\partial \mathcal{I}^{d_I} \partial \mathbf{s}^{d_I}} \mathbf{c}^{d_I} \frac{\partial^2 \widehat{\chi}^{d_I}}{\partial \mathbf{s}^{d_I} \partial \mathcal{I}^{d_I}} - \frac{\partial^2 \widehat{\chi}^{d_I}}{\partial \mathcal{I}^{d_I 2}} \right] \odot \frac{\partial \phi^{d_I}}{\partial Q^{d_I}},\end{aligned}\quad (2.53)$$

after using (2.49) and (2.51). The damage multiplier  $\gamma^{d_I}$  is obtained by imposing damage consistency during persistent damage, as follows.

i. *Damaged unloading.* No further damage of the material occurs in this case, with

$$\phi^{d_I} < 0 \quad \Longrightarrow \quad \gamma^{d_I} = 0, \quad (2.54)$$

by the Kuhn-Tucker condition (2.28)<sub>3</sub>. Therefore, equation (2.49) leads to

$$\boxed{\dot{\mathbf{s}}^{d_I} = \mathbf{c}^{d_I} \dot{\mathbf{e}}^{d_I} \quad \Longrightarrow \quad \mathbf{c}_{lu}^{d_I} = \mathbf{c}^{d_I} \quad \text{for (damaged) unloading,}} \quad (2.55)$$

with no further evolution of the compliance associated to the damage mechanism in this case.

ii. *Damage loading.* The imposition of the consistency condition (2.29)

$$\dot{\phi}^{d_I} = \frac{\partial \phi^{d_I}}{\partial Q^{d_I}} \odot \dot{Q}^{d_I} = 0 \quad (2.56)$$

leads in combination with (2.53) to

$$\gamma^{d_I} = \frac{1}{\Delta^{d_I}} \left( \frac{\partial \phi^{d_I}}{\partial Q^{d_I}} \odot \frac{\partial^2 \widehat{\chi}^{d_I}}{\partial \mathcal{I}^{d_I} \partial \mathbf{s}^{d_I}} \right) \mathbf{c}^{d_I} \dot{\mathbf{e}}^{d_I} \quad (2.57)$$

where

$$\Delta^{d_I} := \frac{\partial \phi^{d_I}}{\partial Q^{d_I}} \odot \left[ \frac{\partial^2 \widehat{\chi}^{d_I}}{\partial \mathcal{I}^{d_I} \partial \mathbf{s}^{d_I}} \mathbf{c}^{d_I} \frac{\partial^2 \widehat{\chi}^{d_I}}{\partial \mathbf{s}^{d_I} \partial \mathcal{I}^{d_I}} - \frac{\partial^2 \widehat{\chi}^{d_I}}{\partial \mathcal{I}^{d_I 2}} \right] \odot \frac{\partial \phi^{d_I}}{\partial Q^{d_I}}. \quad (2.58)$$

which is assumed to be positive  $\Delta^{d_I} > 0$ ; see Remark 2.4.1 below. The introduction of this expression of  $\gamma^{d_I}$  in equation (2.49) leads to the tangent rate equation (2.48) with

$$\boxed{\mathbf{c}_{lu}^{d_I} = \mathbf{c}^{d_I} - \frac{1}{\Delta^{d_I}} \left[ \mathbf{c}^{d_I} \left( \frac{\partial^2 \widehat{\chi}^{d_I}}{\partial \mathbf{s}^{d_I} \partial \mathcal{I}^{d_I}} \odot \frac{\partial \phi^{d_I}}{\partial Q^{d_I}} \right) \right] \otimes \left[ \mathbf{c}^{d_I} \left( \frac{\partial^2 \widehat{\chi}^{d_I}}{\partial \mathbf{s}^{d_I} \partial \mathcal{I}^{d_I}} \odot \frac{\partial \phi^{d_I}}{\partial Q^{d_I}} \right) \right]} \quad (2.59)$$

The symmetry of this tangent is to be noted.

**Remark 2.4.** The constitutive assumption  $\Delta^{d_I} > 0$  in (2.58) is to be verified on a case by case basis in the model examples described in the next section. With this assumption, equation (2.57) leads to the equivalent condition for persistent damage loading

$$\left( \frac{\partial \phi^{d_I}}{\partial Q^{d_I}} \odot \frac{\partial^2 \widehat{\chi}^{d_I}}{\partial \mathcal{I}^{d_I} \partial \mathcal{S}^{d_I}} \right) \mathbf{c}^{d_I} \dot{\boldsymbol{\varepsilon}}^{d_I} > 0, \quad (2.60)$$

for the damage mechanism  $d_I$ , after noting the constraint  $\gamma^{d_I} > 0$  in (2.28)<sub>1</sub>.  $\square$

### 2.4.2. The global elasto-plastic-damage tangent

The above considerations focused on a given damage mechanism  $d_I$ . Global relations between the rates of total strain  $\dot{\boldsymbol{\varepsilon}}$  and stress  $\dot{\boldsymbol{\sigma}}$  can be obtained as follows. The rate form of the elastic relation (2.13) leads to

$$\dot{\boldsymbol{\sigma}} = \mathbf{C}^e \dot{\boldsymbol{\varepsilon}}^e = \mathbf{C}^e (\dot{\boldsymbol{\varepsilon}} - \dot{\boldsymbol{\varepsilon}}^d - \dot{\boldsymbol{\varepsilon}}^p), \quad \text{for } \mathbf{C}^e := \frac{\partial^2 W^e}{\partial \boldsymbol{\varepsilon}^e{}^2} \quad (2.61)$$

the symmetric elastic tangent (not necessarily constant). The plastic strain rates  $\boldsymbol{\varepsilon}^p$  in this last expression are eliminated in the usual way. Indeed, the imposition of the plastic consistency (2.32) in combination with (2.61) leads to the expression of the plastic multiplier

$$\gamma^p = \frac{1}{\Delta^p} \mathbf{n}_{\phi^p} : \mathbf{C}^e (\dot{\boldsymbol{\varepsilon}} - \dot{\boldsymbol{\varepsilon}}^d), \quad (2.62)$$

where we have introduced the notation

$$\mathbf{n}_{\phi^p} := \frac{\partial \phi^p}{\partial \boldsymbol{\sigma}} \quad \text{and} \quad \Delta^p := \mathbf{n}_{\phi^p} : \mathbf{C}^e \mathbf{n}_{\phi^p} + \left( \frac{\partial \phi^p}{\partial Q^p} \right)^2 K^p, \quad (2.63)$$

for the hardening modulus  $K^p := \partial^2 \mathcal{H}^p / \partial \alpha^{p^2}$  in terms of the plastic hardening potential  $\mathcal{H}^p$ . Combining (2.30)<sub>1</sub>, (2.61) and (2.62), we arrive at the tangent relation

$$\dot{\boldsymbol{\sigma}} = \mathbf{C}^{ep} (\dot{\boldsymbol{\varepsilon}} - \dot{\boldsymbol{\varepsilon}}^d), \quad (2.64)$$

for the elastoplastic tangent  $\mathbf{C}^{ep}$

$$\mathbf{C}^{ep} = \mathbf{C}^e - \frac{1}{\Delta^p} \mathbf{C}^e \mathbf{n}_{\phi^p} \otimes \mathbf{C}^e \mathbf{n}_{\phi^p}, \quad (2.65)$$

a symmetric tangent in the assumed associated case.

The combination of equations (2.20)<sub>2</sub> and (2.64) leads to

$$\dot{s}^{d_I} = \mathbb{P}^{d_I} : \dot{\sigma} = \mathbb{P}^{d_I} : \mathbf{C}^{ep} \left( \dot{\epsilon} - \sum_{d_J=1}^{n_{dam}} \mathbb{P}^{d_J T} \dot{e}^{d_J} \right) \quad \text{for } d_I = 1, n_{dam}, \quad (2.66)$$

which with (2.48)<sub>1</sub> leads to

$$\sum_{d_J=1}^{n_{dam}} \left[ \mathbf{c}_{lu}^{d_I} \delta_{d_I d_J} + \mathbb{P}^{d_I} : \mathbf{C}^{ep} \mathbb{P}^{d_J T} \mathbf{c}_{lu}^{d_J} \right] \dot{e}^{d_J} = \mathbb{P}^{d_I} : \mathbf{C}^{ep} \dot{\epsilon} \quad \text{for } d_I = 1, n_{dam}. \quad (2.67)$$

Following the notation introduced in (2.8), we consider the matrix

$$\mathbf{A} := \underbrace{\left[ \mathbf{c}_{lu}^{d_I} \delta_{d_I d_J} + \mathbb{P}^{d_I} : \mathbf{C}^{ep} \mathbb{P}^{d_J T} \right]}_{\text{block } \mathbf{A}_{d_I d_J} \in \mathbb{R}^{n_{d_I} \times n_{d_J}}} \in \mathbb{R}^{n_d^{total} \times n_d^{total}}, \quad (2.68)$$

where

$$n_d^{total} = \sum_{d_I=1}^{n_{dam}} n_{d_I}, \quad (2.69)$$

the total number of damage variables. The symmetry of the matrix  $\mathbf{A}$  is to be noted. From (2.67), we conclude that

$$\dot{e}^{d_I} = \sum_{d_J=1}^{n_{dam}} (\mathbf{A}^{-1})_{d_I d_J} \mathbb{P}^{d_J} : \mathbf{C}^{ep} \dot{\epsilon}, \quad (2.70)$$

where  $(\mathbf{A}^{-1})_{d_I d_J} \in \mathbb{R}^{n_{d_I} \times n_{d_J}}$  denotes the block of the inverse of matrix  $\mathbf{A}$  associated to the mechanisms  $d_I$  and  $d_J$ .

The final tangent rate equation is obtained as

$$\dot{\sigma} = \mathbf{C}^{ep} \left( \dot{\epsilon} - \sum_{d_I=1}^{n_{dam}} \mathbb{P}^{d_I T} \dot{e}^{d_I} \right) \quad (2.71)$$

$$= \left[ \mathbf{C}^{ep} - \sum_{d_I, d_J=1}^{n_{dam}} \mathbf{C}^{ep} : \mathbb{P}^{d_I T} (\mathbf{A}^{-1})_{d_I d_J} \mathbb{P}^{d_J} : \mathbf{C}^{ep} \right] \dot{\epsilon}, \quad (2.72)$$

identifying the final rate equation

$$\dot{\sigma} = \mathbf{C}^{epd} \dot{\epsilon} \quad \text{where} \quad \mathbf{C}^{epd} = \mathbf{C}^{ep} - \sum_{d_I, d_J=1}^{n_{dam}} \mathbf{C}^{ep} : \mathbb{P}^{d_I T} (\mathbf{A}^{-1})_{d_I d_J} \mathbb{P}^{d_J} : \mathbf{C}^{ep}. \quad (2.73)$$

We note the symmetry of the final tangent. The damage incorporated in the tangent response of the material is apparent in the final relation (2.73) through the evolution of the tangent stiffness of the material from its elastoplastic value  $\mathbf{C}^{ep}$ . The degradation of this tangent response is accomplished in the proposed formulation through a direct modeling of the physical damage mechanisms. To observe better the added compliance added to the global tangent response of the material by these mechanisms, we present next an alternative derivation of the tangent in terms of the damage compliances.

### 2.4.3. Alternative form of the damage tangent

An alternative form of the tangent (2.73) can be obtained by deriving first the corresponding compliance, when possible. We first note that the compliance  $\mathbf{d}_{lu}^{d_I} = \mathbf{c}_{lu}^{d_I^{-1}}$  associated to the tangent  $\mathbf{c}_{lu}^{d_I}$  in (2.59) for the damage mechanism  $d_I$  can be obtained using the Sherman-Morrison formula (see e.g. GOLUB & VAN LOAN [1989], page 51), leading to the expression

$$\mathbf{d}_{lu}^{d_I} = \mathbf{d}^{d_I} + \frac{1}{\tilde{\Delta}^{d_I}} \left( \frac{\partial^2 \hat{\chi}^{d_I}}{\partial \mathbf{s}^{d_I} \partial \mathcal{I}^{d_I}} \odot \frac{\partial \phi^{d_I}}{\partial \mathcal{Q}^{d_I}} \right) \otimes \left( \frac{\partial^2 \hat{\chi}^{d_I}}{\partial \mathbf{s}^{d_I} \partial \mathcal{I}^{d_I}} \odot \frac{\partial \phi^{d_I}}{\partial \mathcal{Q}^{d_I}} \right), \quad (2.74)$$

where

$$\begin{aligned} \tilde{\Delta}^{d_I} &= \Delta^{d_I} - \frac{\partial \phi^{d_I}}{\partial \mathcal{Q}^{d_I}} \odot \left[ \frac{\partial^2 \hat{\chi}^{d_I}}{\partial \mathcal{I}^{d_I} \partial \mathbf{s}^{d_I}} \mathbf{c}^{d_I} \frac{\partial^2 \hat{\chi}^{d_I}}{\partial \mathbf{s}^{d_I} \partial \mathcal{I}^{d_I}} \right] \odot \frac{\partial \phi^{d_I}}{\partial \mathcal{Q}^{d_I}} \\ &= -\frac{\partial \phi^{d_I}}{\partial \mathcal{Q}^{d_I}} \odot \frac{\partial^2 \hat{\chi}^{d_I}}{\partial \mathcal{I}^{d_I 2}} \odot \frac{\partial \phi^{d_I}}{\partial \mathcal{Q}^{d_I}}, \end{aligned} \quad (2.75)$$

assuming  $\tilde{\Delta}^{d_I} \neq 0$ ; see Remark 2.5 below. Expression (2.74) can be obtained alternatively by combining (2.52) (instead of (2.53)) and the consistency condition (2.56). For the case of no further damage (i.e. damaged unloading) characterized by the tangent relation (2.55), the corresponding compliance is simply obtained as

$$\mathbf{d}_{lu}^{d_I} = \mathbf{d}^{d_I}, \quad (2.76)$$

given by (2.50).

The global rate equation is then easily obtained from the inverse of the elastic rate equation (2.71) in combination with the inverse of the rate equation (2.48)<sub>1</sub>, that is,

$$\mathbf{D}^{ep} \dot{\boldsymbol{\sigma}} = \dot{\boldsymbol{\varepsilon}} - \underbrace{\sum_{d_I=1}^{n_{dam}} \left( \sum_{a=1}^{n_{d_I}} \mathbb{P}_a^{d_I} \dot{\boldsymbol{\varepsilon}}_a^{d_I} \right)}_{\dot{\boldsymbol{\varepsilon}}^{d_I}} \quad \text{and} \quad \dot{\boldsymbol{\varepsilon}}_a^{d_I} = \sum_{b=1}^{n_{d_I}} \mathbf{d}_{lu\ ab}^{d_I} \underbrace{\mathbb{P}_a^{d_I} : \dot{\boldsymbol{\sigma}}}_{\dot{\boldsymbol{\varepsilon}}_a^{d_I}}, \quad (2.77)$$

for the elastoplastic compliance  $\mathbf{D}^{ep} := \mathbf{C}^{ep^{-1}}$ , thus leading to the elasto-plastic-damage compliance

$$\dot{\boldsymbol{\epsilon}} = \mathbf{D}^{epd} \dot{\boldsymbol{\sigma}} \quad \text{where} \quad \mathbf{D}^{epd} = \mathbf{D}^{ep} + \sum_{d_I=1}^{n_{dam}} \mathbf{D}_{lu}^{d_I}, \quad (2.78)$$

with the damage compliance contribution

$$\mathbf{D}_{lu}^{d_I} = \sum_{a,b=1}^{n_{d_I}} d_{lu\ ab}^{d_I} \mathbb{P}_a^{d_I} \otimes \mathbb{P}_b^{d_I}, \quad (2.79)$$

for the damage mechanism  $d_I$ . The final expression (2.78) identifies clearly the added compliance to the tangent material response

**Remark 2.5.** The compliance (2.74) is only well-defined when the elastoplastic tangent  $\mathbf{C}^{ep}$  is invertible and  $\tilde{\Delta}^{d_I} \neq 0$  in (2.75) (otherwise the tangent  $\mathbf{c}_{lu}^{d_I}$  in (2.59) is singular). After observing the nature of  $\tilde{\Delta}^{d_I}$  in (2.75), we note that this situation may occur in the case of perfect damage (no hardening/softening response of the material), as it is shown in the model examples considered in the next section. The same considerations apply to the invertibility of the elastoplastic tangent (2.65). For these reasons, the form of the tangent (2.73), not requiring this inversion before adding the elastic contributions, is to be preferred in the numerical implementation.  $\square$

### 3. Model Examples

We consider in this section a generic elasto-plastic-damage model involving a quadratic damage potential, and thus introducing a secant damage compliance. Secant damage compliance have been considered in ORTIZ [1985], leading to similar developments to the ones presented herein for the purely elastic-damage case only. After describing this generic quadratic damage model in Section 3.1 in the general elasto-plastic-damage framework developed in the previous section, we consider in Section 3.2 several existing models, developing their formulation within the proposed framework. A single damage mechanism (i.e.  $n_{dam} = 1$ ) is considered in this section.

#### 3.1. A generic quadratic damage model

We consider a damage mechanism characterized by the generic quadratic potential

$$\widehat{W}^d(\mathbf{e}^d; \mathcal{I}^d) = \frac{1}{2} \mathbf{e}^d : \mathbf{c}^d \mathbf{e}^d + \mathcal{H}^d(\alpha^d), \quad (3.1)$$

with the internal variables

$$\mathcal{I}^d = \{ \mathbf{d}^d, \alpha^d \} \quad \text{for} \quad \mathbf{d}^d := \mathbf{c}^{d^{-1}}, \quad (3.2)$$

and for the scalar variable  $\alpha^d$ . The generic function  $\mathcal{H}^d(\alpha^d)$  accounts for an isotropic hardening/softening law of the material. The damage space  $\mathcal{V}^d$  defined in terms of the projection matrices  $\mathbb{P}^d$  by (2.4), so the damage strains are  $\boldsymbol{\varepsilon}^d = \mathbb{P}^{dT} \mathbf{e}^d$ , is to be specified for a particular damage surface; see examples below. We also write the global compliance

$$\mathbf{D}^d = \sum_{a,b=1}^{n_d} d_{ab}^d \mathbb{P}_a^d \otimes \mathbb{P}_b^d, \quad (3.3)$$

and the global tangent

$$\mathbf{C}^d = \sum_{a,b=1}^{n_{d_I}} c_{ab}^d \mathbb{P}_a^d \otimes \mathbb{P}_b^d \quad \text{for} \quad c_{ab}^d := \mathbb{P}_a^d : \mathbf{C}^d \mathbb{P}_b^d. \quad (3.4)$$

leading to the equality

$$\boldsymbol{\varepsilon}^d : \mathbf{C}^d \boldsymbol{\varepsilon}^d = \mathbf{e}^{d_I} \cdot \mathbf{c}^d \mathbf{e}^{d_I}, \quad (3.5)$$

for the quadratic term in (3.1). We write  $\mathbf{C}^{d^{-1}} = \mathbf{D}^d$  in  $\mathcal{V}^d$ , that is, the generalized inverse.

The complementary energy associated to  $\widehat{W}^d$  in (3.1) is given by

$$\begin{aligned} \widehat{\chi}^d(\mathbf{s}^d; \mathbf{d}^d, \alpha^d) &= \frac{1}{2} \mathbf{s}^d : \mathbf{d}^d \mathbf{s}^d - \mathcal{H}^d(\alpha^d) \\ &= \frac{1}{2} (\mathbf{s}^d \otimes \mathbf{s}^d) : \mathbf{d}^d - \mathcal{H}^d(\alpha^d) \end{aligned} \quad (3.6)$$

$$= \frac{1}{2} (\boldsymbol{\sigma}^d \otimes \boldsymbol{\sigma}^d) : \mathbf{D}^d - \mathcal{H}^d(\alpha^d) \quad (3.7)$$

for  $\boldsymbol{\sigma}^d = \mathbb{P}^{dT} \mathbf{s}^d$ , as a calculation based on (2.39) shows. The constitutive relations (2.44) and (2.46) read in this case

$$\mathbf{e}^d = \frac{\partial \widehat{\chi}^d}{\partial \mathbf{s}^d} = \mathbf{d}^d \mathbf{s}^d, \quad (\text{or, equivalently, } \boldsymbol{\varepsilon}^d = \mathbf{D}^d \boldsymbol{\sigma}^d = \mathbf{D}^d \boldsymbol{\sigma}), \quad (3.8)$$

$$\mathcal{Q}_1^d = \frac{\partial \widehat{\chi}^d}{\partial \mathbf{d}^d} = \frac{1}{2} (\mathbf{s}^d \otimes \mathbf{s}^d), \quad (3.9)$$

$$\mathcal{Q}_2^d = \frac{\partial \widehat{\chi}^d}{\partial \alpha^d} = -\frac{\partial \mathcal{H}^d}{\partial \alpha^d} =: q^d, \quad (3.10)$$

leading to the expression (2.16) of the damage dissipation

$$\mathcal{D}^d = \frac{1}{2} (\mathbf{s}^d \otimes \mathbf{s}^d) : \dot{\mathbf{d}}^d + q^d \dot{\alpha}^d \geq 0. \quad (3.11)$$

Appropriate damage surfaces are defined then in this case as

$$\phi^d(\frac{1}{2}(\mathbf{s}^d \otimes \mathbf{s}^d), q^d) \leq 0, \quad (3.12)$$

leading to the associated damage evolution equations

$$\dot{\mathbf{d}}^d = \gamma^d \mathbf{N}_{\phi^d} \quad \text{for} \quad \mathbf{N}_{\phi^d} := \frac{\partial \phi^d}{\partial (\frac{1}{2}(\mathbf{s}^d \otimes \mathbf{s}^d))}, \quad (3.13)$$

$$\dot{\alpha}^d = \gamma^d \frac{\partial \phi^d}{\partial q^d}, \quad (3.14)$$

with the Kuhn-Tucker loading/unloading conditions (2.28) and the consistency conditions (2.29).

**Remark 3.1.** The constraint (3.11) on the damage dissipation  $\mathcal{D}^d$  is automatically satisfied assuming a damage surface of the form

$$\phi^d(\frac{1}{2}\mathbf{s}^d \otimes \mathbf{s}^d, q) = g^d(\underbrace{\frac{1}{2}\mathbf{s}^d \otimes \mathbf{s}^d}_{\mathcal{Q}_1^d}) - \underbrace{(y_o^d - q^d(\alpha^d))}_{y^d(\alpha^d) \geq 0} \leq 0 \quad (3.15)$$

for a material parameter  $y^d(0) = y_o^d > 0$  (i.e.  $q^d(0) = 0$ ) and a function  $g^d(\cdot)$  positively homogeneous of degree 1. We recall that a positively homogeneous function of degree  $m \geq 1$  is defined by

$$g^d(\lambda \mathcal{Q}_1^d) = \lambda^m g^d(\mathcal{Q}_1^d) \quad \forall \lambda \in \mathbb{R}^+ \quad \implies \quad \frac{\partial g^d}{\partial \mathcal{Q}_1^d} : \mathcal{Q}_1^d = m g^d, \quad (3.16)$$

the Euler's theorem of homogeneous functions. In the case of interest ( $m = 1$ ), we obtain after some straightforward algebraic manipulations

$$\mathcal{D}^d = \frac{1}{2}(\mathbf{s}^d \otimes \mathbf{s}^d) : \dot{\mathbf{d}}^d + q^d \dot{\alpha}^d = \gamma^d y_o^d \geq 0 \quad (3.17)$$

given the Kuhn-Tucker condition (2.28). □

### 3.1.1. An alternative characterization of the damage

The damage evolution equations (3.13)-(3.14) have been obtained, following the general framework presented in Section 2, in terms of the damage surface (3.12) function of the fourth order tensor  $\mathcal{Q}_1^d = \frac{1}{2}(\mathbf{s}^d \otimes \mathbf{s}^d)$ . It is common, however, to characterize the damage of the material in terms of a damage surface of the form

$$\tilde{\phi}^d(\mathbf{s}^d, q^d) \leq 0 \quad (3.18)$$

in terms of the stress tensor  $\mathbf{s}^d$ . The damage evolution equations for this case can still be obtained using the principle of maximum damage dissipation as follows.

Consider the Lagrangian associated to the dissipation functional (3.11) and the constraint introduced by the damage surface (3.18), that is,

$$\mathcal{L}^d(\mathbf{s}^d, q^d, \gamma^d; \dot{\mathbf{e}}^d, \dot{\alpha}^d) := \frac{1}{2} \mathbf{s}^d \cdot \dot{\mathbf{d}}^d \mathbf{s}^d + q^d \dot{\alpha}^d - \gamma^d \tilde{\phi}^d(\tilde{\mathbf{s}}^d, q^d), \quad (3.19)$$

for a Lagrange multiplier  $\gamma^d \geq 0$  imposing the unilateral constraint  $\tilde{\phi}^d \leq 0$ . The maximization of the Lagrangian (3.19) for fixed rates  $\dot{\mathbf{d}}^d$  and  $\dot{\alpha}^d$  of the damage variables leads to the evolution equations

$$\dot{\mathbf{d}}^d \mathbf{s}^d = \gamma^d \mathbf{n}_{\tilde{\phi}^d} \quad \text{and} \quad \dot{\alpha}^d = \gamma^d \frac{\partial \tilde{\phi}^d}{\partial q^d} \quad (3.20)$$

with the Kuhn-Tucker complementary conditions (2.28), where we have introduced the notation

$$\mathbf{n}_{\tilde{\phi}^d} := \frac{\partial \tilde{\phi}^d}{\partial \mathbf{s}^d}. \quad (3.21)$$

Clearly, the only component of the compliance rate affecting the dissipation functional (3.19) is the compliance along the direction defined by the current damage stress  $\mathbf{s}^d$ . Hence, the relation (3.20) shows that the principle of maximum dissipation in combination of a damage surface in terms of the stress  $\mathbf{s}^d$  alone determines only this component of the compliance, leaving arbitrary the component in the direction orthogonal (in stress space) to the current stress. A particular evolution equation for the compliance  $\mathbf{d}^d$  satisfying the relation (3.20)<sub>1</sub> is given by

$$\dot{\mathbf{d}}^d = \gamma^d \frac{\mathbf{n}_{\tilde{\phi}^d} \otimes \mathbf{n}_{\tilde{\phi}^d}}{\mathbf{n}_{\tilde{\phi}^d} \cdot \mathbf{s}^d}, \quad (3.22)$$

as long as  $\mathbf{n}_{\tilde{\phi}^d} \cdot \mathbf{s}^d \neq 0$ ; see Remark 3.2.2 below. Evolution equations of the form given by (3.22) can be found in the works of SIMO et al [1993] and GOVINDJEE, KAY & SIMO [1995] in the context of damage models based on the evolution of the a total compliance.

The two different expressions (3.13) and (3.22) can be reconciled by noting that the consideration of the function

$$\tilde{\phi}^d(\mathbf{s}^d, q^d) := \phi\left(\frac{1}{2}(\mathbf{s}^d \otimes \mathbf{s}^d), q^d\right), \quad (3.23)$$

for a given function  $\phi^d(\cdot)$ , leads to the differential relation

$$\begin{aligned} d\tilde{\phi}^d &= \mathbf{n}_{\tilde{\phi}^d} \cdot d\mathbf{s}^d + \frac{\partial \tilde{\phi}^d}{\partial q^d} dq^d \\ &= \frac{\mathbf{n}_{\tilde{\phi}^d} \otimes \mathbf{n}_{\tilde{\phi}^d}}{\mathbf{n}_{\tilde{\phi}^d} \cdot \mathbf{s}^d} : \underbrace{\frac{1}{2}[d\mathbf{s}^d \otimes \mathbf{s}^d + \mathbf{s}^d \otimes d\mathbf{s}^d]}_{d\left(\frac{1}{2}\mathbf{s}^d \otimes \mathbf{s}^d\right)} + \frac{\partial \tilde{\phi}^d}{\partial q^d} dq^d \end{aligned}$$

$$\begin{aligned}
&= d\phi^d = \frac{\partial\phi^d}{\partial(\frac{1}{2}\mathbf{s}^d \otimes \mathbf{s}^d)} : d(\frac{1}{2}\mathbf{s}^d \otimes \mathbf{s}^d) + \frac{\partial\phi^d}{\partial q^d} dq^d \\
&= \frac{\partial\phi^d}{\partial(\frac{1}{2}\mathbf{s}^d \otimes \mathbf{s}^d)} \mathbf{s}^d \cdot d\mathbf{s}^d + \frac{\partial\phi^d}{\partial q^d} dq^d,
\end{aligned} \tag{3.24}$$

implying

$$\mathbf{n}_{\bar{\phi}^d} = \frac{\partial\phi^d}{\partial(\frac{1}{2}\mathbf{s}^d \otimes \mathbf{s}^d)} \mathbf{s}^d = \mathbf{s}^d \frac{\partial\phi^d}{\partial(\frac{1}{2}\mathbf{s}^d \otimes \mathbf{s}^d)}, \tag{3.25}$$

and  $\partial\phi^d/\partial q = \partial\bar{\phi}^d/\partial q$ . Hence, the evolution formulas (3.13) and (3.22) define the same (maximum) damage dissipation along the direction of the current stress  $\mathbf{s}^d$ . For future use and in general, we write the damage evolution equation

$$\dot{\mathbf{d}}^d = \gamma^d \mathbf{N}_{\phi^d}^d, \tag{3.26}$$

with  $\mathbf{N}_{\phi^d}$  given by (3.13) or (3.22). The general relation

$$\mathbf{n}_{\bar{\phi}^d} = \mathbf{N}_{\phi^d} \mathbf{s}^d = \mathbf{s}^d \mathbf{N}_{\phi^d} \tag{3.27}$$

is also satisfied in both cases by (3.25). Given this relation, we can express the rate of the damage strains as

$$\dot{\mathbf{e}}^d = \mathbf{d}^d \dot{\mathbf{s}}^d + \dot{\mathbf{d}}^d \mathbf{s}^d = \mathbf{d}^d \dot{\mathbf{s}}^d + \gamma^d \mathbf{n}_{\bar{\phi}^d} \tag{3.28}$$

or, equivalently,

$$\dot{\mathbf{s}}^d = \mathbf{c}^d (\dot{\mathbf{e}}^d - \gamma^d \mathbf{n}_{\bar{\phi}^d}), \tag{3.29}$$

involving the current secant damage tensor  $\mathbf{c}^d = \mathbf{d}^{d-1}$ . We note that the ‘‘degrading strains’’ (rates), referred to in the introduction, correspond to the last term in (3.28). We conclude then that it is the first term on the right-hand side of (3.28), which is also active in unloading, which gives the recoverable character of the damage strains  $\mathbf{e}^d$ .

The damage tangent associated to this generic quadratic damage mechanism is also obtained using the equations (2.55) and (2.59), leading to the expressions

**i. Damaged unloading:**

$$\mathbf{c}_{lu}^d = \mathbf{c}^d, \tag{3.30}$$

**ii. Loading (persistent damage):**

$$\mathbf{c}_{lu}^d = \mathbf{c}^d - \frac{1}{\Delta^d} \mathbf{c}^d \mathbf{n}_{\bar{\phi}^d} \otimes \mathbf{c}^d \mathbf{n}_{\bar{\phi}^d}, \tag{3.31}$$

for

$$\Delta^d = \underbrace{\mathbf{n}_{\bar{\phi}^d} : \mathbf{c}^d \mathbf{n}_{\bar{\phi}^d}}_{\tilde{\Delta}^d} + K^d \left( \frac{\partial\phi^d}{\partial q^d} \right)^2 \quad \text{where} \quad K^d := \frac{d^2 \mathcal{H}^d}{d\alpha^{d^2}}, \tag{3.32}$$

the hardening/softening tangent modulus. We note the nature of  $\tilde{\Delta}^d$  as discussed in Section 3.1.1.

The final global tangent is obtained using the general formula (2.73). We note again that the damage space basis  $\mathbb{P}^d$  is to be defined below for the cases of interest.

### Remarks 3.2.

1. Similarly to the arguments presented in Remark 3.1, the constraint on the damage dissipation  $\mathcal{D}^d \geq 0$  is automatically enforced by starting directly from a damage surface of the form

$$\tilde{\phi}^d(\mathbf{s}^d, q^d) = \tilde{g}^d(\mathbf{s}^d) - \underbrace{(y_o^d - q^d(\alpha^d))}_{\tilde{y}^d(\alpha^d) \geq 0} \leq 0 \quad (3.33)$$

for a positively homogeneous function  $\tilde{g}^d(\cdot)$  of degree  $m$ , so

$$\frac{\partial \tilde{g}^d}{\partial \mathbf{s}^d} \cdot \mathbf{s}^d = \mathbf{n}_{\tilde{\phi}^d} \cdot \mathbf{s}^d = m \tilde{g}^d(\mathbf{s}^d). \quad (3.34)$$

Combining (3.11) with (3.22), we obtain after some straightforward algebraic manipulations

$$\mathcal{D}^d = \frac{m}{2} \gamma^d \left[ y_o^d - \left(1 - \frac{2}{m}\right) q^d \right] \quad (3.35)$$

Therefore, we conclude  $\mathcal{D}^d \geq 0$  for a homogeneous function of degree  $m = 2$ . For the case of a homogeneous function of degree  $m = 1$ , a positive dissipation is also concluded if  $q^d \geq -y_o^d$ ; in particular, for the case of a softening law  $q^d \geq 0$ .

2. Regarding the formula (3.22) for the evolution of the damage compliance  $\mathbf{d}^d$ , we note that the denominator is non-zero; in fact

$$\mathbf{n}_{\tilde{\phi}^d} \cdot \mathbf{s}^d > 0 \quad \text{for} \quad \tilde{\phi}^d(\mathbf{s}^d, q^d) = 0 \quad \text{and} \quad \mathbf{s}^d \neq 0, \quad (3.36)$$

if  $\tilde{\phi}^d$  defines a convex elastic damage domain in the space of damage stresses  $\mathbf{s}^d$  ( $q^d = \text{constant}$ ) including the origin  $\mathbf{s}^d = 0$ , as it is usually the case. In fact, for the common case (3.33) with  $\tilde{g}^d(\cdot)$  homogeneous of degree  $m > 0$ , we have

$$\mathbf{n}_{\tilde{\phi}^d} \cdot \mathbf{s}^d = m \tilde{g}^d(\mathbf{s}^d a) = m \tilde{y}^d(\alpha^d) > 0 \quad (3.37)$$

as long as  $0 \leq \tilde{y}^d(\alpha^d) \neq 0$ , that is, when the elastic domain does not reduce to  $\mathbf{s}^d \equiv 0$  (or, in other words, when it has not shrunk to the origin due to a softening of the material). In this situation, the relation (3.22) reflects the physical fact  $\mathbf{d}^d \rightarrow \infty$  for

a fully damaged material; note that  $\mathbf{c}^d \rightarrow 0$  as needed in the expressions (2.55) and (2.59) of the tangent.  $\square$

### 3.2. Some existing models

The generic quadratic model developed in the previous section is fully defined once the damage surfaces (3.23), defining in the process the damage space (2.4) and the nature of the damage compliance (3.2), are defined. With specific particular definitions of these surfaces, the considered general formulation encompasses the models developed in ORTIZ [1985] and SIMO & JU [1987], and later considered in HANSEN & SCHREYER [1994], SIMO et al [1993] and GOVINDJEE, KAY & SIMO [1995] in the modeling of the damage in brittle materials, concrete in particular. We show in the next sections that other existing stress-based damage models can be obtained in this way. We consider for simplicity the case of a linear elastic material, that is, a quadratic stored energy function

$$W^e(\boldsymbol{\varepsilon}^e) = \frac{1}{2} \boldsymbol{\varepsilon}^e : \mathbf{C}^e \boldsymbol{\varepsilon}^e, \quad (3.38)$$

for a constant  $\mathbf{C}^e$ , leading to the quadratic complementary energy function

$$\chi^e(\boldsymbol{\sigma}) = \frac{1}{2} \boldsymbol{\sigma} : \mathbf{D}^e \boldsymbol{\sigma} = \frac{1}{2} (\boldsymbol{\sigma} \otimes \boldsymbol{\sigma}) : \mathbf{D}^e, \quad (3.39)$$

for  $\mathbf{D}^e = \mathbf{C}^{e^{-1}}$ .

#### 3.2.1. An effective stress anisotropic damage model

The formulation of anisotropic damage models can be often found in the literature developed after the introduction of the so-called damage tensors defining an effective stress tensor. Early examples are the works of CORDEBOIS & SIDOROFF [1982] and MURAKAMI [1983]. In the context developed in the previous section, we first note the secant relation

$$\begin{aligned} \boldsymbol{\varepsilon} - \boldsymbol{\varepsilon}^p &= \boldsymbol{\varepsilon}^e + \boldsymbol{\varepsilon}^d = \left[ \mathbf{D}^e + \mathbf{D}^d \right] \boldsymbol{\sigma} \\ &= \mathbf{D}^e \left[ \mathbb{I} + \mathbf{C}^e \mathbf{D}^d \right] \boldsymbol{\sigma}, \end{aligned} \quad (3.40)$$

for the four order identity tensor  $\mathbb{I}$ . Hence, we can write

$$\bar{\boldsymbol{\sigma}} := \mathbf{C}^e (\boldsymbol{\varepsilon} - \boldsymbol{\varepsilon}^p) = \mathbf{M}^{-1} \boldsymbol{\sigma} \quad \text{for} \quad \mathbf{M} := \left[ \mathbb{I} + \mathbf{C}^e \mathbf{D}^d \right]^{-1}, \quad (3.41)$$

and the *effective stresses*  $\bar{\boldsymbol{\sigma}}$ . Numerous anisotropic damage models has been formulated through the consideration of a general effective stress tensor  $\mathbf{M}$ ; we refer to LEMAITRE & CHABOCHE [1978,85], CORDEBOIS & SIDOROFF [1982] and MURAKAMI [1983], among others.

The framework developed in the previous section identifies an evolution for this effective damage tensor  $\mathbf{M}$ . Namely, using the evolution equation (3.26), we obtain

$$\dot{\mathbf{M}} = -\gamma^d \mathbf{M} \mathbf{C}^e \mathbf{N}_{\phi^d} \mathbf{M} , \quad (3.42)$$

for  $\mathbf{N}_{\phi^d}$  given by (3.13) or (3.22) for a general damage function  $\phi^d(\frac{1}{2}(\boldsymbol{\sigma} \otimes \boldsymbol{\sigma}), q^d) \leq 0$  or  $\tilde{\phi}^d(\boldsymbol{\sigma}, q^d) \leq 0$ , respectively. More importantly, this evolution equation is obtained in the thermodynamic framework developed in the previous sections and, in particular, through the direct physical modeling of the damage mechanism through the stress/damage strain relations characterizing it. This situation is to be contrasted with usual ad-hoc characterizations of the evolution of the effective stress tensor commonly found in the literature.

**Remark 3.3.** Having defined the effective stresses  $\bar{\boldsymbol{\sigma}}$  in terms of the damage tensor  $\mathbf{M}$  in (3.41), one may consider the yield criterion (2.26)<sub>2</sub> in the effective stress space, that is,

$$\bar{\phi}^p(\bar{\boldsymbol{\sigma}}, q^p) \leq 0 , \quad (3.43)$$

as it is commonly found in the literature; see e.g. HANSEN & SCHREYER [1994] and JU [1989]. This option, however, leads to a non-symmetric tangent.  $\square$

### 3.2.2. An isotropic damage model

Damage models in the form of a scalar damage variable characterizing the isotropic damage of the material can be traced to the pioneering work of KACHANOV [1958]. Isotropic damage can be characterized by the consideration damage surface

$$\phi^d(\boldsymbol{\sigma}, q^d) = \chi^e(\boldsymbol{\sigma}) - (y_o^d - q) \leq 0 , \quad (3.44)$$

in terms of the elastic complementary energy function (3.39). An equivalent expression (using twice the square root of  $\chi^e$ ) can be found in SIMO & JU [1987]. The threshold value to activate the damage is denoted by  $y_o^d$  and evolves with the scalar hardening/softening variable  $q$ , given in terms of the conjugate strain-like variable  $\alpha^d \in [0, \infty)$  by (3.17) for the hardening/softening potential  $\mathcal{H}^d(\alpha^d)$ . The appearance of all the components of the stress  $\boldsymbol{\sigma}$  (not specific components  $\boldsymbol{\sigma}^d = \mathbb{P}^{d^T} \mathbf{s}^d$  only) in the damage surface (3.44) identifies the associated damage space with the full space of symmetric tensors, that is,

$$\mathcal{V}^d \equiv \mathcal{S} . \quad (3.45)$$

In this situation, we can choose  $\mathbf{e}^d \equiv \boldsymbol{\varepsilon}^d$  (and  $\mathbf{s}^d \equiv \boldsymbol{\sigma}^d = \boldsymbol{\sigma}$ ), with no need to introduce the projections  $\{\mathbb{P}_a^d\}_{a=1}^{n_d}$  in (2.6) (or, in other words, they are the identity). Similarly, we work with the global damage compliance  $\mathbf{D}^d \equiv \mathbf{d}^d$  in this case.

The damage evolution equations (3.13) read in this case

$$\dot{\mathbf{D}}^d = \gamma^d \mathbf{D}^e, \quad (3.46)$$

$$\dot{\alpha}^d = \gamma^d, \quad (3.47)$$

for the elastic compliance  $\mathbf{D}^e$ . For the particular case (3.39) involving a constant  $\mathbf{D}^e$ , the damage evolution equations (3.46)-(3.47) can be easily integrated to arrive at the expression

$$\mathbf{D}^d = \int \dot{\mathbf{D}} dt = \int \dot{\alpha}^d dt \mathbf{D}^e = \alpha^d \mathbf{D}^e. \quad (3.48)$$

The damage strains are easily obtained using (2.44) with the complementary function (3.6), leading to

$$\boldsymbol{\varepsilon}^d = \mathbf{D}^d \boldsymbol{\sigma} = \alpha^d \mathbf{D}^e \boldsymbol{\sigma} = \alpha^d \boldsymbol{\varepsilon}^e, \quad (3.49)$$

thus resulting in the relations

$$\boldsymbol{\varepsilon}^e = \frac{1}{(1 + \alpha^d)} (\boldsymbol{\varepsilon} - \boldsymbol{\varepsilon}^p) = (1 - d) (\boldsymbol{\varepsilon} - \boldsymbol{\varepsilon}^p), \quad (3.50)$$

$$\boldsymbol{\varepsilon}^d = \frac{\alpha^d}{1 + \alpha^d} (\boldsymbol{\varepsilon} - \boldsymbol{\varepsilon}^p) = d (\boldsymbol{\varepsilon} - \boldsymbol{\varepsilon}^p), \quad (3.51)$$

after introducing the definition

$$\boxed{d := \frac{\alpha^d}{1 + \alpha^d} \implies d \in [0, 1]}. \quad (3.52)$$

In this notation, we can write

$$\mathbf{D}^d = \frac{d}{1 - d} \mathbf{D}^e, \quad (3.53)$$

for the damage secant compliance  $\mathbf{D}^d$ . The stress is given by

$$\boldsymbol{\sigma} = \mathbf{C}^e \boldsymbol{\varepsilon}^e = (1 - d) \mathbf{C}^e (\boldsymbol{\varepsilon} - \boldsymbol{\varepsilon}^p) = (1 - d) \bar{\boldsymbol{\sigma}} \quad \text{for} \quad \bar{\boldsymbol{\sigma}} := \mathbf{C}^e (\boldsymbol{\varepsilon} - \boldsymbol{\varepsilon}^p), \quad (3.54)$$

with  $\mathbf{C}^e = \mathbf{D}^{e^{-1}}$ . Relations (3.50) and (3.54) define the classical concepts of effective strains and stresses, respectively, in the isotropic case, as it can be found in the classical literature on the subject (see the complete account in KACHANOV [1986]). Equation (3.54) is a particular case of the general anisotropic relation (3.41), with  $\mathbf{M} = (1 - d)\mathbf{I}$ .

The damage tangent moduli  $\mathbf{C}_{lu}^d$  given by (3.30)-(3.31) reads in this case

i. Damaged unloading:

$$\mathbf{C}_{lu}^d = \frac{(1 - d)}{d} \mathbf{C}^e, \quad (3.55)$$

ii. Damage loading:

$$\mathbf{C}_{lu}^d = \frac{(1-d)}{d} \left[ \mathbf{C}^e - \frac{1}{\bar{\boldsymbol{\sigma}} : \mathbf{D}^e \bar{\boldsymbol{\sigma}} + \frac{K^d d}{(1-d)^3}} \bar{\boldsymbol{\sigma}} \otimes \bar{\boldsymbol{\sigma}} \right], \quad (3.56)$$

Finally, the total tangent moduli

$$\dot{\boldsymbol{\sigma}} = \mathbf{C}^{epd} \dot{\boldsymbol{\varepsilon}}, \quad (3.57)$$

given by the general formula (2.73), reads in this case of a single isotropic damage mechanism

i. Damaged unloading:

$$\mathbf{C}^{epd} = (1-d) \mathbf{C}^{ep}, \quad (3.58)$$

ii. Damage loading:

$$\mathbf{C}^{epd} = (1-d) \left[ \mathbf{C}^{ep} - \frac{1}{\bar{\boldsymbol{\sigma}} : \mathbf{D}^e \bar{\boldsymbol{\sigma}} + \frac{K^d d}{(1-d)^4}} \bar{\boldsymbol{\sigma}} \otimes \bar{\boldsymbol{\sigma}} \right], \quad (3.59)$$

for the isotropic hardening/softening modulus  $K^d$  defined in (3.32)<sub>2</sub>. The case of elastic damage is recovered when  $\mathbf{C}^{ep} = \mathbf{C}^e$ .

### 3.2.3. Smeared crack models

The modeling of cracking in concrete has been often formulated in the framework of the so-called smeared crack models, especially in the context of finite element analyses. Representative early applications of this approach can be found in RASHID [1968], BAZANT & CEDOLIN [1979] and ROTS et al [1985], among many others. Despite its popularity, the lack of a sound thermodynamic framework as opposed to continuum damage models has, perhaps, made this approach less attractive from the theoretical point of view. We show in this section that smeared crack models fit perfectly in the thermodynamic framework of continuum damage models encompassed by the simple generic quadratic model presented above.

For a given a nucleation criterion (say Rankine's maximum stress criterion for brittle materials), a single crack can be characterized by its normal direction, the unit vector  $\mathbf{n}$ . The cohesive opening/closing of the crack is then assumed controlled by the traction vector associate to  $\mathbf{n}$ , that is,

$$\mathbf{T} = \boldsymbol{\sigma} \mathbf{n}. \quad (3.60)$$

Let  $\{\mathbf{m}_1 \equiv \mathbf{n}, \dots, \mathbf{m}_{n_{\text{dim}}}\}$  denote an orthonormal Cartesian system in a general  $n_{\text{dim}}$  dimensional setting (so  $\mathbf{m}_i \cdot \mathbf{m}_j = \delta_{ij}$  for  $i, j = 1, n_{\text{dim}}$ ). Denoting the corresponding components of the traction vector  $\mathbf{T}$  by

$$s_a^d \equiv T_a := \mathbf{m}_a \cdot \mathbf{T} \quad \text{for } a = 1, n_{\text{dim}}, \quad (3.61)$$

the general damage surface (3.23) in terms of the reduced damage stresses  $\mathbf{s}^d$  corresponds in this case to a damage surface in terms of the traction vector on the crack. These considerations identify the damage space  $\mathcal{V}^d$  for the damage mechanism associated to a single crack. Namely, the projection matrices into this space are given by

$$\mathbb{P}_a^d = \frac{1}{2} (\mathbf{m}_a \otimes \mathbf{n} + \mathbf{n} \otimes \mathbf{m}_a) \quad \text{for } a = 1, n_{\text{dim}}, \quad (3.62)$$

(so  $\mathbb{P}_1^d = \mathbf{n} \otimes \mathbf{n}$ , in particular). The orthogonality relation (2.5) can be easily verified. The damage strains are then constructed by (2.4) in terms of the so-called crack strains  $\mathbf{e}^d$ , with  $n_d = n_{\text{dim}}$  in this case.

The damage evolution equation (3.26) reads in this case

$$\dot{\mathbf{d}}^d = \gamma^d \mathbf{N}_{\phi^d}, \quad (3.63)$$

for the reduced crack compliance  $\mathbf{d}^d \in \mathbb{R}^{n_{\text{dim}} \times n_{\text{dim}}}$ , a rank two (not four) tensor in this case, with  $\mathbf{N}_{\phi^d}$  given by (3.13) or (3.22), in terms of a damage surface involving the components of the traction vector  $\mathbf{T}$ . We refer to KROPLIN & WEIHE [1997], among many others, for the determination of such damage surfaces for the modeling of cracking in concrete.

**Remark 3.4.** The inviscid smeared crack models considered in this section in combination of strain softening are known to lead to fundamental difficulties, the well-known pathological mesh dependence of the resulting finite element solutions in particular. To avoid these inconsistencies, we have presented in ARMERO [1997a,b] an alternative formulation of this type of anisotropic damage models in the context of the so-called strong discontinuities. This approach considers the limit solutions with a discontinuous displacement field; see e.g. SIMO et al [1993], and ARMERO & GARIKIPATI [1996]. More recently, we have presented in ARMERO [1997a,99] a multi-scale framework for the introduction of these solutions, or better the corresponding localized dissipative mechanisms, in the large scale problem of the local continuum. The corresponding strains in the small scale (a local neighborhood of a point  $\mathbf{x}$ ) are given

$$\boldsymbol{\varepsilon}_\mu = \bar{\boldsymbol{\varepsilon}}_\mu + (\boldsymbol{\xi} \otimes \mathbf{n})^s \delta_{\Gamma_x}, \quad (3.64)$$

for the Dirac delta function  $\delta_{\Gamma_x}$  associated to the discontinuity surface  $\Gamma_x$  with unit normal  $\mathbf{n}$ , with a regular part of the local strain  $\bar{\boldsymbol{\varepsilon}}_\mu$  (determining the stress  $\boldsymbol{\sigma}$ ) and a jump displacement field  $\boldsymbol{\xi}$  across  $\Gamma_x$ . The decomposition (3.64) fits along the developments presented in this section with  $\boldsymbol{\varepsilon}^e \equiv \bar{\boldsymbol{\varepsilon}}_\mu$  and  $\boldsymbol{\varepsilon}^d = \mathbb{P}^{dT} \mathbf{e}^d$  with  $\mathbb{P}_a^d$  defined as in (3.63) and  $\mathbf{e}_a^d = \boldsymbol{\xi}_a \delta_{\Gamma_x}$  ( $a = 1, n_{\text{dim}}$ ). The damage potential  $W^d(\cdot)$  is given in this case by

$$\widehat{W}^d(\mathbf{e}^d; \mathcal{I}^d) = \widetilde{W}^d(\boldsymbol{\xi}; \widetilde{\mathcal{I}}^d) \delta_{\Gamma_x}, \quad (3.65)$$

for a set of localized internal variables  $\mathcal{I}^d = \tilde{\mathcal{I}}^d \delta_{\Gamma_x}$ . Taking the differential of (3.65) we obtain

$$d\widehat{W}^d = \frac{\partial \widetilde{W}^d}{\partial \boldsymbol{\xi}} \cdot \underbrace{d(\boldsymbol{\xi} \delta_{\Gamma_x})}_{\mathbf{e}^d} + \frac{\partial \widetilde{W}^d}{\partial \tilde{\mathcal{I}}^d} \odot \underbrace{d(\tilde{\mathcal{I}}^d \delta_{\Gamma_x})}_{\mathcal{I}^d} \implies \begin{cases} \mathbf{s}^d = \frac{\partial \widetilde{W}^d}{\partial \boldsymbol{\xi}}, \\ \mathcal{Q}^d = -\frac{\partial \widetilde{W}^d}{\partial \tilde{\mathcal{I}}^d}, \end{cases} \quad (3.66)$$

being both regular distributions. For the quadratic model (3.1), the relation (3.65) leads to  $\mathbf{d}^d = \tilde{\mathbf{d}}^d \delta_{\Gamma_x}$  and  $\alpha^d = \tilde{\alpha}^d \delta_{\Gamma_x}$ , with the secant relation (3.8) and the constitutive relation (3.10) reading in this case

$$\boldsymbol{\xi} = \tilde{\mathbf{d}}^d \mathbf{s}^d \quad \text{and} \quad q^d = -\frac{\partial \tilde{\mathcal{H}}^d}{\partial \tilde{\alpha}^d}, \quad (3.67)$$

for a localized softening potential  $\tilde{\mathcal{H}}^d(\tilde{\alpha}^d)$ . The dissipation (3.11) is then given by

$$\mathcal{D}^d = \underbrace{\left[ \frac{1}{2}(\mathbf{s}^d \otimes \mathbf{s}^d) : \dot{\tilde{\mathbf{d}}}^d + q^d \dot{\tilde{\alpha}}^d \right]}_{\tilde{\mathcal{D}}^d} \delta_{\Gamma_x}, \quad (3.68)$$

reflecting its localized nature. The damage evolution equations (3.26) read now

$$\dot{\tilde{\mathbf{d}}}^d = \tilde{\gamma}^d \mathbf{N}_{\phi^d} \quad \text{and} \quad \dot{\tilde{\alpha}}^d = \tilde{\gamma}^d \frac{\partial \phi^d}{\partial q^d}, \quad (3.69)$$

for  $\mathbf{N}_{\phi^d}$  defined as in (3.13) or (3.22), and a localized damage consistency parameter localized on  $\Gamma_x$  (with  $\gamma^d = \tilde{\gamma}^d \delta_{\Gamma_x}$ ), satisfying similarly the Kuhn-Tucker loading-unloading conditions (2.28) and consistency conditions (2.29). The damage surfaces  $\phi^d$  are defined as considered above for smeared crack models, but now resulting directly the stress-displacement (3.67)<sub>1</sub> relations, with associated compliance  $\tilde{\mathbf{d}}^d$ . These relations are directly introduced in the finite element solutions through a local enhancement of the finite elements; we refer to the aforementioned references for complete details.  $\square$

## 4. Concluding Remarks

We have presented in this first paper a general framework for the formulation of continuum damage models in the infinitesimal range. This framework is based on the kinematic assumption of the strains decomposed in elastic, plastic and damage parts, similar to the elastoplastic decompositions of the strains in plasticity models. The inclusion in this assumption of a general thermodynamic framework describing the response of the material leads then to a complete characterization of the damage in the material. Most notably,

these considerations allow for a complete and independent characterization of the damage mechanisms, in contrast with more traditional approaches found in the literature based on the effective stress concept through the evolution of the so-called “damage tensors”. The resulting formulation leads then to a very modular treatment of the damage effects in solids, requiring only the physically motivated modeling of the strain/stress response associated to the damage mechanism independently of the overall response of the solid. In addition, we have also shown that classical models, from isotropic damage models to smeared crack models, fit in the proposed formulation, thus giving a must needed unified framework to these existing approaches.

In conclusion, this work has identified a framework leading to a physically motivated modeling of plastic damage in materials with an efficient numerical treatment. Even though the concept of damage strain as used herein can be found in the literature as noted in the introduction, most notably through the so-called “crack strains” in the modeling of damage in concrete, their use in the formulation of continuum damage models seem not to be widely spread. We believe that the highly modular structure of the proposed formulation clearly justifies its consideration, even in the case when the formulation reduces to existing models. This is especially the case when developing numerical schemes for the integration of the resulting constitutive models, as we present in Part II of this work in combination with a simple plastic damage model of porous metals. It is our intend to continue this work with the consideration of additional couplings between the different inelastic mechanisms (see Remark 3.3), as well as the extension of these ideas to the finite deformation range.

## Acknowledgments

Financial support for this research has been provided by the ONR under contract no. N00014-96-1-0818 with UC Berkeley. This support is gratefully acknowledged. We also acknowledge the support of the NSF under contract no. CMS-9703000 with UC Berkeley.

## References

- ARMERO, F. [1997a] "On the Characterization of Localized Solutions in Inelastic Solids: an Analysis of Wave Propagation in a Softening Bar," to appear in *Comp. Meth. Appl. Mech. Eng.* (available also as UCB/SEMM Report no. 97/18)
- ARMERO, F. [1997b] "Localized Anisotropic Damage of Brittle Materials," *Proc. COMPLAS V*, eds. D.R.J. Owen, E. Onate, and E. Hinton, CIMNE, Barcelona.
- ARMERO, F. [1999] "Large-Scale Modeling of Localized Dissipative Mechanisms in a Local Continuum: Applications to the Numerical Simulation of Strain Localization in Rate-Dependent Inelastic Solids," *Mechanics of Cohesive-Frictional Materials*, **4**, 101-131.
- ARMERO, F. & GARIKIPATI, K. [1996] "An Analysis of Strong Discontinuities in Multiplicative Finite Strain Plasticity and their Relation with the Numerical Simulation of Strain Localization in Solids," *Int. J. Solids and Structures*, **33**, 2863-2885.
- ARMERO, F. & LI, S. "Numerical Simulation of Anisotropic Damage at Finite Strains," *Proc. 12th ASCE Eng. Mech. Conf.*, La Jolla, CA, May 1998.
- ARMERO, F. & OLLER, S. [1999] "A General Framework for Continuum Damage Models. Part II: Applications to the Numerical Simulation of Porous Metals," submitted to *Int. J. Solids and Structures*.
- BAZANT & CEDOLIN [1979] "Blunt Crack Band Propagation in Finite Element Analysis," *Journal of the Engineering Mechanics Division*, ASCE, **105**, 297-315.
- BAZANT, Z. & OH, J. [1983] "Crack Band Theory for Fracture of Concrete," *Materials and Structures*, RILEM, **16**, 155-177.
- BAZANT, Z. & OH, B.H. [1985] "Microplane Model for Progressive Fracture of Concrete and Rock," *J. Eng. Mech.*, **111**, 559-582.
- CAROL, I.; RIZZI, E. & WILLAM, K. [1994] "Unified Theory of Elastic Degradation and Damage Based on a Loading Surface," *Int. J. Solids Struct.*, **31**, 2835-2865.
- CHABOCHE, J.L. [1995] "A Continuum Damage Theory with Anisotropic and Unilateral Damage," *La Recherche Aéronautique*, **2**, 139-147.
- CORDEBOIS, J. & SIDOROFF, F. [1982] "Endommagement anisotrope", *J. Mécanique Theor. Appl.*, spécial.
- DOUGILL, J.W. [1976] "On stable progressively fracturing solids," *ZAMP*, **27**, 423-37.
- GOLUB G.H. & VAN LOAN, C.F. [1989] *Matrix Computations*, Second Edition, The John Hopkins University Press, Baltimore.

- GOVINDJEE, S.; KAY, G.J. & SIMO, J.C. [1995] "Anisotropic Modeling and Numerical Simulation of Brittle Damage in Concrete," *Int. J. Num. Meth. Eng.*, **38**, 3611-33.
- HANSEN, N.R. & SCHREYER, H.L. [1994] "A Thermodynamically Consistent Framework for Theories of Elastoplasticity Coupled with Damage," *Int. J. Solids Struct.*, **31**, 359-389.
- HUECKEL, T. & MAIER, G. [1977] "Incrementally Boundary-Value Problems in the Presence of Coupling of Elastic and Plastic Deformations: a Rock Mechanics Oriented Theory," *Int. J. Solids Structure*, **13**, 1-15
- JU, J.W. [1989] "On energy-based coupled elastoplastic damage theories: constitutive modeling and computation aspects," *Int. J. Solids Structures*, **25**, 803-833.
- KACHANOV, L.M. [1958] "Time Rupture Process under Creep Conditions," (in Russian), *Izv. ARad. SSSR Teckh. Nauk.*, **8**, 26-31
- KACHANOV, L.M. [1986] *Introduction to Continuum Damage Mechanics*, Martinus Nijhoff Publishers, Dordrecht.
- KROPLIN, B. & WEIHE, S. [1997] "Constitutive and Geometrical Aspects of Fracture Induced Anisotropy," *Proc. COMPLAS V*, eds. D.R.J. Owen, E. Onate, and E. Hinton, CIMNE, Barcelona.
- LEMAITRE, J. & CHABOCHE, J.L. [1978] "Aspect Phénoménologique de la Rupture per Endommagement," *J. Mécanique Appliquée*, **2.**, 317-365.
- LEMAITRE, J. & CHABOCHE, J.L. [1985] *Mécanique de Matériaux Solide*, Dunod, Paris.
- LUBLINER [1990] *Plasticity Theory*, MacMillan, New York.
- LUBLINER, J.; OLIVER, J.; OLLER, S. & OÑATE, E. [1989] "A Plastic-Damage Model for Concrete," *Int. J. Solids and Structures*, **25**, 299-326.
- LUCCIONI, B.; OLLER, S. & DANESI, R. [1996] "Coupled Plastic-Damage Model" *Comp. Meth. Appl. Mech. Eng.*, **129**, 81-89.
- MURAKAMI, S. [1983] "Notion of Continuum Damage Mechanics and its Application to Anisotropic Creep Damage Theory," *J. Eng. Mat. Techn.* **105**, 99-105.
- ORTIZ, M. [1985] "A Constitutive Theory for the Inelastic Behavior of Concrete," *Mech. Mat.*, **4**, 67-93.
- ORTIZ, M. [1987] "Analytical Study of the Localized Failure Modes of Concrete." *Mechanics of Materials*, *Mech. Mat.*, **6**, 159-174.
- RASHID, Y.R. [1968] "Analysis of Prestressed Concrete Pressure Vessels," *Nuclear Engineering and Design*, **7**, 334-344.

- 
- ROTS, J.G.; NAUTA, P.; KUSTERS, G. & BLAAUWENDRAA, T. [1985] "Smearred Crack Approach and Fracture Localization in Concrete," *Heron*, **30**.
- SIMO, J.C. & HUGHES, T.J.R. [1998] *Computational Inelasticity*, Springer Verlag.
- SIMO, J.C. & JU, J.W. [1987] "Strain- and Stress-Based Continuum Damage Models. Part I: Formulation. Part II: Computational Aspects," *Int. J. Solids Structures*, **23**, 821-840 and 841-869.
- SIMO, J.C; OLIVER, J. & ARMERO, F. [1993] "An Analysis of Strong Discontinuities Induced by Softening Solutions in Rate Independent Solids," *J. Comput. Mech.*, **12**, 277-296.
- TRUESDELL & NOLL [1965] "The Nonlinear Field Theories of Mechanics," *Handbich der Physik Bd. III/3*, ed. by S. Fluegge, Springer Verlag, Berlin.
- YAZADANI, S. & SCHREYER, H.L. [1988] "An Anisotropic Damage Model with Dilatation for Concrete," *Mechanics of Materials*, **7**, 231-244.